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# Wavelength Modulation Spectroscopy for Temperature and Species Concentration in the Plume of a Supersonic Nozzle

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We measure the plume temperature and species concentration profiles of a heated mixture of ammonia and steam flowing through a miniature converging-diverging nozzle (area ratio 5:1, exit diameter 4.57 mm). The nozzle operating conditions are intended to simulate the plume of a thruster system operating in vacuum environments. Wavelength modulation spectroscopy (WMS) is a laser absorption spectroscopy technique that allows for quantitative, time-resolved, sensitive diagnostics of gaseous flows. The goal of the measurements is to assess the potential of WMS as a diagnostic technique for characterizing this type of system. The measurements use three multiplexed fiber-coupled tunable diode lasers: two for measuring temperature and H<sub>2</sub>O concentration and one for measuring NH<sub>3</sub> concentration. The multiplexed laser beam is spatially translated across the plume on a two-axis stage. Using Abel transforms, we are able to successfully resolve the structure of the under-expanded plume despite absorption signals as low as 9e-4. As a result, WMS demonstrates excellent promise as a sensitive diagnostic tool for resolving the profile of such plumes.

## Nomenclature

|                 |   |  |
|-----------------|---|--|
| $A_\lambda$     | = | path averaged absorbance area of feature at wavelength, $\lambda$                  |
| $\chi_i$        | = | mole fraction of species, $i$  |
| $I_0$           | = | incident intensity on gas sample   |
| $I_t$           | = | transmitted intensity out of gas sample  |
| $L$             | = | pathlength probed by laser sensor  |
| $\lambda$       | = | wavelength of light  |
| $\nu$           | = | optical frequency of light   |
| $\phi(\lambda)$ | = | lineshape function of absorption feature   |
| $S_\lambda(T)$  | = | linestrength of absorption feature   |
| $T$             | = | temperature of gas sample  |
| $R$             | = | outer radius of jet  |
| $r$             | = | radial position within jet   |
| $y$             | = | vertical position within jet   |
| $X_{2f}$        | = | X component of lockin amplifier at $2f$  |
| $Y_{2f}$        | = | Y component of lockin amplifier at $2f$  |
| $R_{1f}$        | = | root-sum-square magnitude of the X and Y components of the lock-in outputs at $1f$ |
| $A_\lambda$     | = | radial distribution of absorbance area of feature at wavelength, $\lambda$         |

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## I. Introduction

Plume diagnostics are important tools for characterizing the operation of rockets and other propulsion systems. However, many systems are difficult to probe due to their size (both large and small), high velocities, caustic products, and varying pressure conditions. Laser based diagnostics are ideal for these environments as they are time-resolved, non-intrusive, and can be very sensitive. Time-resolved measurements of species concentration and plume temperature can be used to validate models and quantify the state-of-health of these systems. Wavelength modulation spectroscopy (WMS) is a method of absorption spectroscopy particularly well suited for these measurements because it is quantitative, reduces noise, and has improved sensitivity over traditional tunable diode laser methods [1]-[2].

In absorption spectroscopy, light is passed through a gas sample and will be absorbed if the wavelength of the light is resonant with a molecular transition. For homogenous mixtures, absorbance ( $\alpha$ ) of the light at a given wavelength ( $\lambda$ ), is quantified by Beer's Law (1).  $I_o$  represents the initial intensity of the light, and  $I_t$  represents the transmitted intensity of the light after it has passed through the sample. From the absorbance, gas properties such as species concentration ( $X_i$ ), pressure ( $P$ ), pathlength ( $L$ ) and temperature (through the lineshape  $S_\lambda(T)$ ) can be calculated, if the lineshape function  $\Phi$  is known (or if an entire isolated lineshape is measured, since  $\int_{-\infty}^{+\infty} \phi(\lambda) d\lambda \equiv 1$ ). Transition specific parameters such as  $S_\lambda(T)$  and  $\Phi$  are calculated from spectroscopic parameters found in databases such as HITRAN 2012 [3] or from experimental studies [4]–[6].

$$\alpha(\lambda) = S_\lambda(T) P X_i L \phi(\lambda) = -\ln \left( \frac{I_t}{I_o} \right) \quad (1)$$

Tunable diode lasers (TDLs) are commonly used in absorption spectroscopy due to their small size, low cost relative to other laser sources, and robust nature (resulting from their development by the telecom industry). The wavelength of a TDL tunes approximately linearly with the tuning of the laser injection current, allowing the center wavelength of the TDL to be swept over an absorption feature [7]. WMS is a modification of tunable diode laser absorption spectroscopy where the wavelength of the TDL is rapidly modulated at a frequency  $f$  (here, approximately 100 kHz) [2]. In the presence of absorption, this modulation gives rise to harmonics of the original modulation frequency ( $f$ ) in the detected intensity signal. The second harmonic ( $2f$ ) is of particular interest, as the magnitude is linearly related to absorbance and can therefore be used to infer thermodynamic properties of the gas. If the second harmonic signal alone is used for thermodynamic measurements, the laser intensity must be calibrated in-situ with a known mixture, which is not possible in many engineering applications due to window fouling, vibrating optics and beam steering. It has been demonstrated that dividing the second harmonic signal by the first harmonic signal removes the dependence on laser intensity, thus eliminating the need for intensity calibrations [1]. The resulting signal is referred to hereafter as the normalized  $2f$  signal. A lock-in amplifier is used to isolate the harmonics, filtering noise outside of the bandwidth of the lock-in amplifier, and shifting the detection to higher frequencies where laser and system noise is decreased. By comparing the shape and magnitude of the normalized  $2f$  signal from the lock-in detector to simulated signals, thermodynamic properties such as pressure, temperature and species concentration can be measured.

## II. Experimental Setup

The WMS measurements are performed at the Air Force Research Laboratory at Edwards Air Force Base in California. The converging-diverging nozzle and heating system was mounted inside of chamber 4 within the Spacecraft Propulsion Laboratory. The laser light and translation stage control system signals were passed into chamber 4 via fiber optic and electrical vacuum feedthroughs on the chamber.

The exhaust plume examined in this study is composed of a mixture of anhydrous ammonia and water. The water concentration of the plume is held at approximately 18% water by mass. The gas flow is preheated by two circulation heaters connected in series. Liquid water from a syringe pump, is injected between the two heating stages, exposing it to the preheated ammonia flow. The syringe pump allows the water flow rate to be precisely controlled and varied. The temperature of the tube where the water enters the ammonia flow stream is monitored by a type K thermocouple. To adequately vaporize the water, water is not injected into the system until the tube, at the entry point, reaches a minimum temperature of 100 °C. To minimize system exposure to the corrosive effects of ammonia, the system is preheated and purged with nitrogen.

Immediately following the second heating stage, the gas mixture flows through the converging-diverging nozzle seen in Figure 1 to create the plume. This nozzle configuration results in an expansion ratio of 5.19 and an exit Mach number of 3.05. By design, this nozzle creates a highly under-expanded plume at the operational background pressures of 550-720 mTorr.

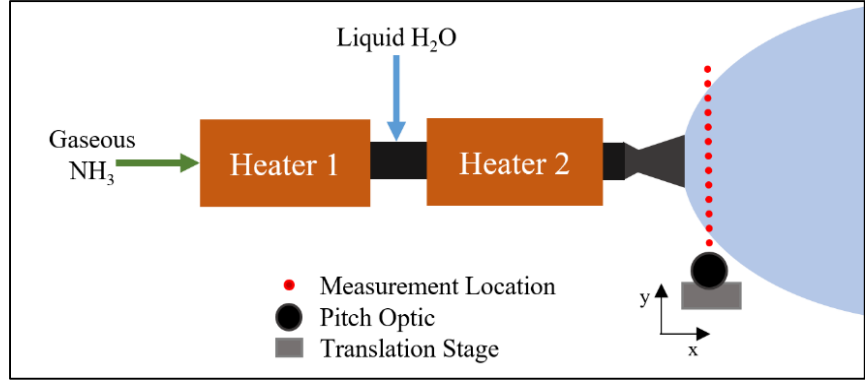


Figure 1: Experimental setup of  $\text{NH}_3$  and  $\text{H}_2\text{O}$  injection with translation stage

Three tunable diode lasers are used to probe three distinct absorption features. Laser 1 (Eblana EP1388-DM) and laser 2 (Eblana EP1392-DM) probe  $\text{H}_2\text{O}$  absorption features at 1388 nm and 1392 nm respectively. Laser 3 (Toptica LD-1470-0010-DFB-1) probes an  $\text{NH}_3$  feature at 1527nm. The transition at 1388nm reaches a maximum absorption at low temperature, while the transition at 1392nm reaches a maximum absorption at elevated temperatures (approximately 650K). The different temperature dependencies of the  $\text{H}_2\text{O}$  absorption features, results in an absorbance ratio of the two lines that is sensitive to temperature, but also results in low peak absorbance at low temperatures for the feature at 1391.7nm.

A National Instruments PXIe-6368 data acquisition card provides the voltage signal used to modulate the laser injection currents for WMS. The light from the three fiber-coupled lasers is multiplexed onto a single-mode fiber and passed into the vacuum chamber through an optical fiber feed-through, to a launch optic (Thorlabs CFC-5X-C). The beam traverses the plume perpendicular to the flow direction, as shown in Figure 1. The beam is collected by a catch collimator (Oz Optics HPUCO-25-1300/1550-M-25AC-SP) to a 400-micron diameter multimode fiber (Thorlabs M45L01) and passes back through the optical feed-through to an InGaAs photodetector (Thorlabs PDA10CS). The signal from the photodetector is monitored by the PXIe-6368 data acquisition card at a 2MHz sampling rate.

The launch and collection optics are mounted on a 2-axis stage that can be translated in the vertical and horizontal directions through the use of two stepper motors. The translation system enables measurements from -0.4 cm to +0.4 cm relative to the plume center at vertical increments of 0.1 mm, resulting in 81 total measurement locations at 1 mm downstream of the nozzle exit.

### III. Data Processing

#### A. Residual Amplitude Modulation

The magnitude of the absorption signal is linearly correlated with the path length and approximately linearly correlated with pressure, as seen in Equation 1. The small exit diameter of the nozzle (4.57mm) and the low operating pressures within the vacuum chamber (approximately 500 mTorr) resulted in low absorption signals.

$$2f/1f = \sqrt{\left[\left(\frac{X_{2f}}{R_{1f}}\right)_{raw} - \left(\frac{X_{2f}}{R_{1f}}\right)_{bg}\right]^2 + \left[\left(\frac{Y_{2f}}{R_{1f}}\right)_{raw} - \left(\frac{Y_{2f}}{R_{1f}}\right)_{bg}\right]^2} \quad (2)$$

Due to the low pressure and short pathlength, the magnitude of the absorption signal is on the same order of magnitude as the residual amplitude modulation (RAM) of the laser [8], [9]. RAM is caused by non-linear intensity tuning of the laser system that creates harmonic signals even in the absence of absorption (WMS background). It is present in varying degrees in all tuned laser systems. Figure 2 shows the normalized  $2f$  signal of the transmitted intensity (i.e. after absorption) in green and the normalized  $2f$  signal of the incident intensity (i.e. without absorption),

in red. Theoretically, if the laser tuning is exactly linear, the normalized  $2f$  of the incident intensity would be constant at zero.

In Figure 2a it is clear that the non-linear tuning creates a strong background in the normalized  $2f$  signal of the transmitted intensity. The incident intensity is recorded concurrently with the transmitted intensity; this allows the second harmonic components of the non-absorbing incident intensity signal to be subtracted from the second harmonic components of the absorbing transmitted intensity signal according to Equation 2. The parameters  $X_{2f}$ ,  $Y_{2f}$  are the X and Y components of the signals from the digital lockin at  $2f$ .  $R_{1f}$  refers to the root mean square of the X and Y parameters at  $1f$ . Finally, the components denoted 'raw' are from the transmitted intensity signal before RAM subtraction, while the components denoted 'bg' or background are from the incident intensity signal. With this procedure, we can isolate the components of the transmitted intensity signal caused by absorption. The background-subtracted signal is shown in Figure 2b. This method is applied to all measured signals for all three lasers. The 1392nm feature, shown in Figure 2, is the weakest absorption signal ( $9E-4$  absorbance, or 0.09% absorption), and therefore the most sensitive to the background signal and subtraction.

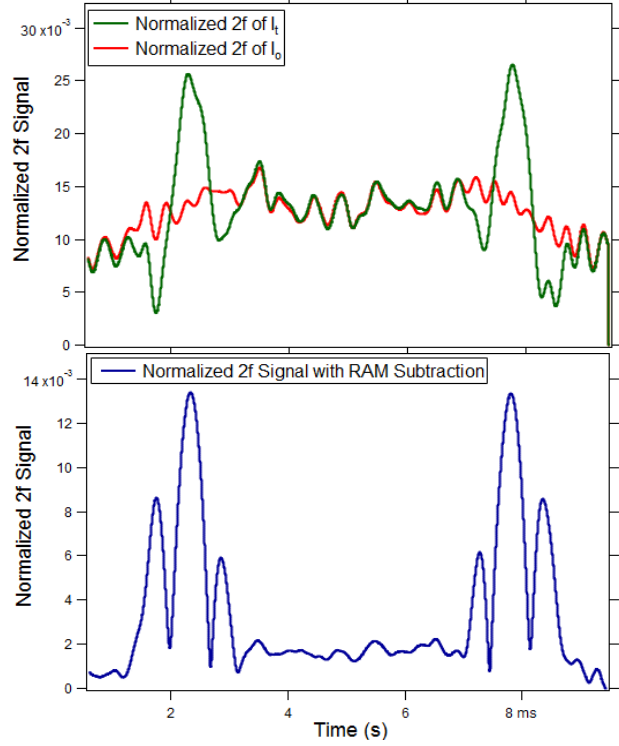


Figure 2: Figure 2a: Normalized  $2f$  signal of transmitted intensity with absorption before RAM subtraction (green), Normalized  $2f$  signal of incident intensity without absorption, showing the RAM contribution (red). Figure 2b: Normalized  $2f$  signal with RAM subtraction.

## B. Averaging

For these measurements, we perform scanned-wavelength WMS, where the high frequency wavelength modulation is combined with an additional slow sine wave modulation (100Hz) to scan the average wavelength of the laser across the entire absorption feature [10]. The basic time resolution of the measurement is therefore 200 Hz, as we scan the full normalized  $2f$  signal for each feature twice per cycle of the slow modulation.

Figure 3 shows a single sweep, and a 0.5s average of the normalized  $2f$  signals. Though the single and 0.5s averages are similar for the stronger features probed by lasers 1 and 3, the 0.5s average significantly improves data analysis for the weak 1392nm feature. For the results reported in the remainder of the paper, 0.5s averages are used.

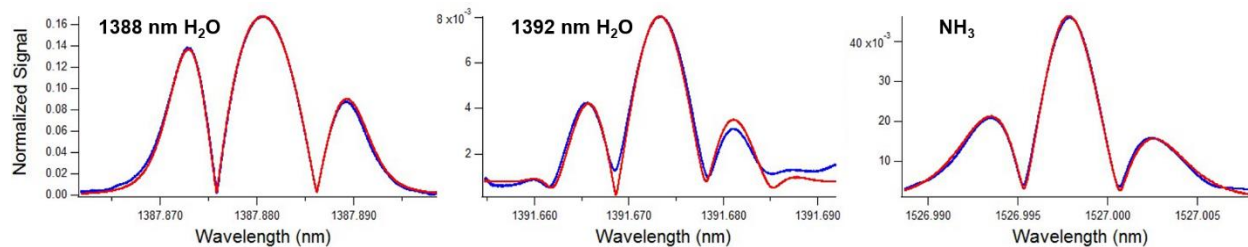


Figure 3: Raw data (red) versus time averaged data (blue) for each absorption feature. Figure 3a: 1388nm  $H_2O$  feature, Figure 3b: 1392nm  $H_2O$  feature, Figure 3c:  $NH_3$  feature.

### C. Absorbance Area

We utilize a fitting procedure similar to that described by [11] to calculate an integrated absorbance area for each absorption feature [6], [3], [5]. Using the integrated absorbance area, rather than the peak normalized 2f signal, simplifies the interpretation of the measured data by removing the dependence of the signal on line broadening. An example of a fit produced with this procedure is shown in Figure 4. The good agreement between the fit and data ensures that the absorbance area calculated by the fit is representative of the absorbance area of the measured absorption feature.

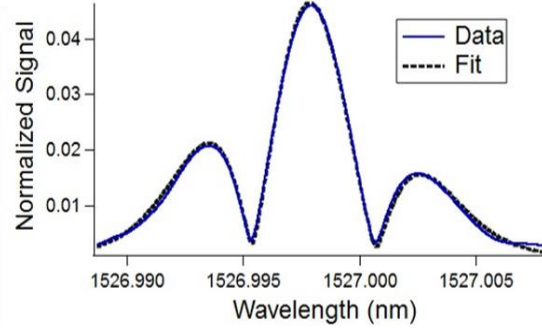


Figure 4: Fitted normalized signals (black) versus time averaged normalized signals (blue) for 1527nm  $\text{NH}_3$  feature at 1mm downstream of nozzle exit.

The fitting procedure is repeated for each measurement location within the 81-point vertical scan at 1 mm downstream of the nozzle exit. The resulting areas for all three absorption features at each point within the scan are displayed in Figure 5 as a function of vertical position of the beam within the plume. The absorbance areas do not decrease to zero at the edges of the plume due to residual water vapor and ammonia within the vacuum chamber. To account for the residual signal, absorbance areas for all three absorption features are measured outside of the plume and are subtracted to isolate absorbance within the plume for temperature and species concentration calculations.

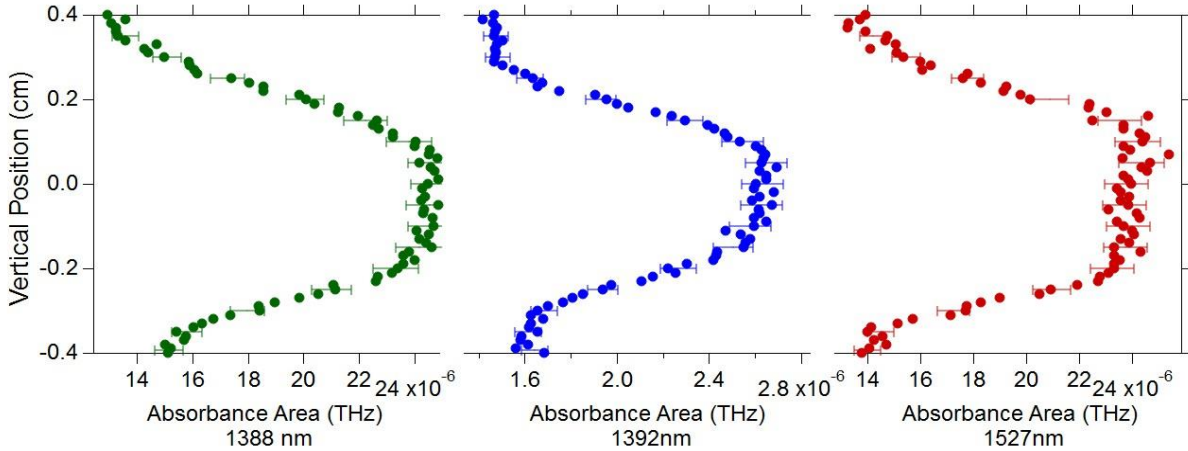


Figure 5: Absorbance areas as a function of vertical position in plume. a) Absorbance area of 1388nm  $\text{H}_2\text{O}$  feature. b) Absorbance area of 1392nm  $\text{H}_2\text{O}$  feature. c) Absorbance area of 1527nm  $\text{NH}_3$  feature.

## IV. Path Averaged Results

### A. Temperature

Two-line thermometry, described by [12], is implemented to calculate the path averaged temperature within the plume. By taking the ratio of the calculated absorbance areas from the two  $\text{H}_2\text{O}$  absorption features, the measurement dependence on pressure, path length and species concentration is removed, as seen in Equation 3. Under these circumstances, the ratio of the absorbance areas,  $A_{1388}$  and  $A_{1392}$ , simplifies to the ratio of the linestrengths of the  $\text{H}_2\text{O}$  absorption features,  $S_{1388}(T)$  and  $S_{1392}(T)$ , which are solely a function of temperature. Two-line thermometry is especially useful for plume measurements, as the plume width ( $L$ ) and pressure distributions are not precisely known at each measurement location.

$$\frac{A_{1388}}{A_{1392}} = \frac{S_{1388}(T)X_{\text{H}_2\text{O}}LP_{\text{total}}}{S_{1392}(T)X_{\text{H}_2\text{O}}LP_{\text{total}}} = \frac{S_{1388}(T)}{S_{1392}(T)} \quad (3)$$

Using this approach, the path-averaged temperature at each point within the plume at 1mm downstream of the nozzle exit plane is calculated and shown in Figure 6. At the edges of the plume the absorbance signal from residual  $H_2O$  and  $NH_3$  make up a larger percentage of the total measured signal, as the absorbance signal contributed from the plume decreases. This causes the uncertainty associated with the calculated background absorbance to be magnified at the edges of the plume as the background becomes a larger percentage of the signal.

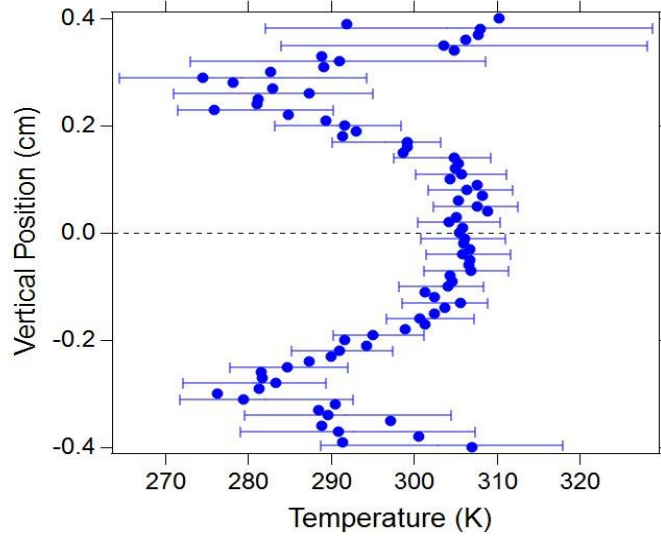


Figure 6: Path averaged temperature profile as a function of vertical position within the plume.

## B. Species Concentration

Common methods for calculating species concentration in the plume are not effective for this application as the pathlength of the laser through the plume and the pressure distribution are unknown. Here we introduce a new method of species concentration measurement that is possible when one is measuring all of the species present in the flow. To begin, we calculate the ratio of the absorbance areas of one of the water absorbance features and the ammonia absorbance feature. Both absorbance areas were recorded at the same path length ( $L$ ) and pressure ( $P_{total}$ ), therefore the dependence on these parameters is removed (Equation 4).

In this system we assume there are no reactions taking place, therefore the only molecules within the system are the ammonia and water vapor. Under this situation, the sum of the mole fractions of ammonia and water are equal to one.

$$\frac{A_{1388}}{A_{1527}} = \frac{S_{1388}(T)X_{H_2O}LP_{total}}{S_{1527}(T)X_{NH_3}LP_{total}} = \frac{S_{1388}(T)X_{H_2O}}{S_{1527}(T)X_{NH_3}} \quad (4)$$

$$\sum_{i=1}^N X_i = 1 \rightarrow X_{NH_3} + X_{H_2O} = 1 \quad (5)$$

$$X_{NH_3} = \frac{S_{1388}(T)A_{1527}}{A_{1388}S_{1527}(T) + A_{1527}S_{1388}(T)} \quad (6)$$



By combining Equations 4 and 5 and solving for the mole fraction of ammonia we arrive at Equation 6. Since the path averaged temperature has already been calculated, the right side of this equation is entirely known, enabling the calculation of the path-averaged mole fractions of the absorbing species at each location in the scan (see Figure 7). Since the vacuum chamber was evacuated prior to trials, we expect the mole fraction of  $\text{NH}_3$  and  $\text{H}_2\text{O}$  to be approximately constant throughout the plume. The variation in mole fractions in Figure 7 is likely due to incomplete mixing of the  $\text{NH}_3$  and  $\text{H}_2\text{O}$  prior to the nozzle, or non-linear temperature effects resulting from path averaged measurements over a range of temperatures. The large uncertainties on the edges of the plume is due to temperature uncertainty, which is reflected in linestrength uncertainty in Equation 6.

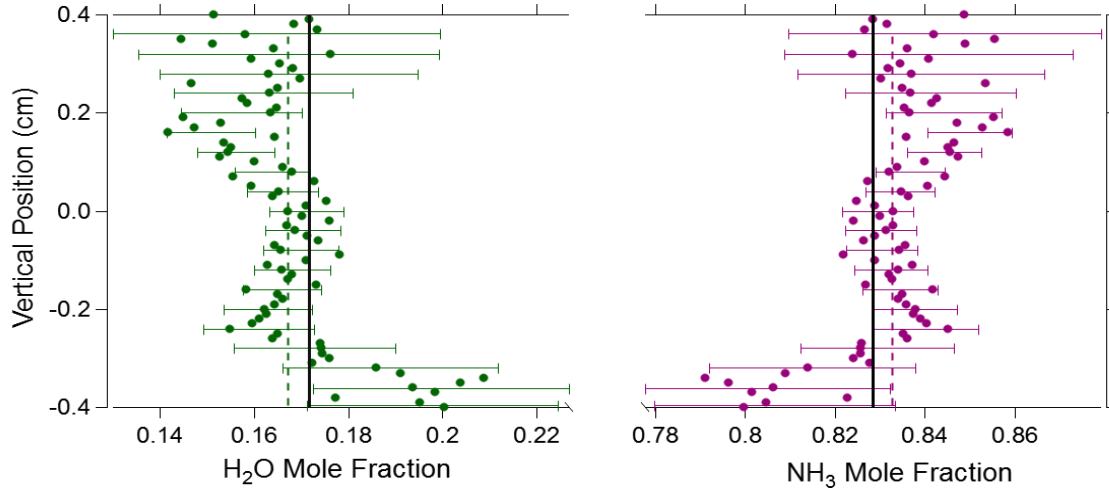


Figure 7: Water and ammonia mole fraction distributions in plume. Dashed lines represent the mean mole fraction of the data. Solid black lines represent the expected mole fraction based on the flow rates of  $\text{H}_2\text{O}$  and  $\text{NH}_3$  into the chamber.

The dashed lines in Figure 7 represent the average mole fractions of  $\text{H}_2\text{O}$  and  $\text{NH}_3$  from the data,  $0.167 \pm 0.013$  and  $0.833 \pm 0.013$  respectively. The solid black lines represent the expected mole fraction based on the flow rates of  $\text{H}_2\text{O}$  and  $\text{NH}_3$  into the chamber, 0.172 and 0.828 respectively. The expected mole fractions fall well within one standard deviation of the calculated values. The agreement between expected values and measured values suggests that the proposed species concentration technique is valid for this application and that all significant constituents of the flow are accounted for.

Discrepancies in the measured mole fractions versus the expected mole fractions could be due to reactions taking place within the system, which we did not account for by assuming the only constituents in the flow were  $\text{H}_2\text{O}$  and  $\text{NH}_3$ , or from water vapor condensing out of the flow due to rapid temperature drops from the rapid expansion into the vacuum chamber. At the elevated temperatures, such as the temperature within the heaters (approximately 800K),  $\text{NH}_3$  will partially dissociate into  $\text{N}_2$  and  $\text{H}_2$  as in Equation 7. While the residence time the ammonia was kept at elevated temperatures was short, this reaction likely decreased the mole fraction of the  $\text{NH}_3$  present in the system. This would also decrease the  $\text{H}_2\text{O}$  mole fraction since the total number of moles in the system is increased by this reaction.



Although condensation is not observed within the system during testing, it is possible that the  $\text{H}_2\text{O}$  vapor is condensing in the flow, particularly at the edges of the plume where the temperatures are the lowest. Condensation would cause the vapor mole fraction of  $\text{H}_2\text{O}$  to decrease, while increasing the vapor mole fraction of  $\text{NH}_3$ . This is consistent with the calculated mole fractions being lower than the expected values in Figure 7.



## V. Abel Transform Results

Given the radially symmetric nozzle geometry and the symmetry of the path averaged measurement results about the center axis of the nozzle exit, we assume that the plume is approximately axisymmetric. This allows the path averaged integrated absorbance measurements ( $A_\lambda(y)$ ) from Figure 5 to be converted into integrated absorbance as a function of radius ( $A'_\lambda(r)$ ), by the use of an inverse Abel transform (Equation 8). This method is described in depth in [13].

$$A'_\lambda(r) = -\frac{1}{\pi} \int_y^R \frac{dA_\lambda(y)}{dy} \frac{dy}{\sqrt{y^2 - r^2}} \quad (8)$$

We can reconstruct the radial temperature profile within the plume using two-line thermometry and the radial absorbance area distributions of the two H<sub>2</sub>O features (Figure 8). From isentropic flow calculations for pure NH<sub>3</sub>, we estimate that the temperature at the core of the plume at the exit of the nozzle should be approximately 350K. As our measurement is downstream of the exit plane, it is logical that the measured temperatures are lower than this predicted exit temperature. Computational fluid dynamics models would be required to determine the expected temperatures at the measurement location. The high temperature core and the low temperature edges of the plume are consistent with the expected nearfield under-expanded structure [14], [15]. As the optics are translated past to the plume boundary, we see the temperature return to near atmospheric temperature.

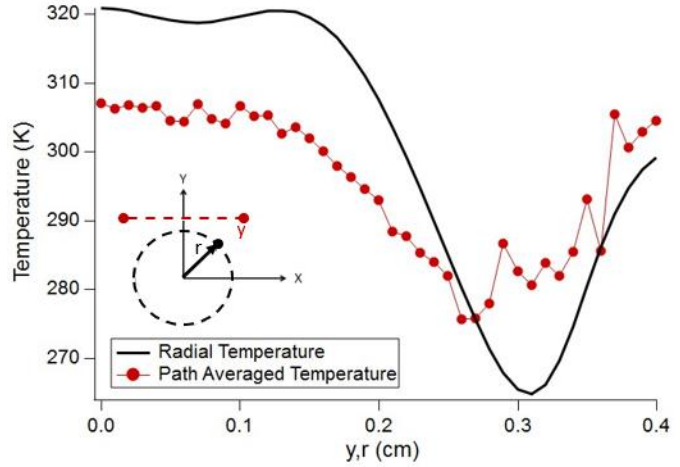


Figure 8: Radial temperature distribution from inverse Abel transform

## VI. Conclusion

We resolve path-averaged and radial profiles of temperature within the plume of a small diameter (<5mm) converging-diverging nozzle expanding into a vacuum chamber at Edwards Air Force Base. A new method of calculating species mole fraction was proposed, for systems in which pressure and path length are uncertain, and for which all species are being measured. While the species mole fraction measurement technique includes some obvious sources of uncertainty, the uncertainties associated with estimating pressure and path length in a system such as this would be far greater. The calculated mole fractions compare well with expected values, indicating this mole fraction calculation technique is valid. Temperature results from path averaged and radially distributions agree with theoretical predictions, but a computational fluid dynamics model would be needed to fully validate this. The positive results from this study indicate that WMS could be applied to operational propulsion systems to gain temporally and spatially resolved measurements within the thruster plumes.

## Acknowledgements

The University of Colorado authors were supported under contract ROS151366C from Jacobs Engineering. Helpful discussions with members of the Precision Laser Diagnostics Laboratory and AFRL In-Space Propulsion Branch are gratefully acknowledged.

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# Wavelength Modulation Spectroscopy for Temperature and Species Concentration in the Plume of a Supersonic Nozzle

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N.A. MacDonald <sup>3</sup>, W.A. Hargus<sup>3</sup> and G.B. Rieker <sup>1</sup>

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Edwards Air Force Base, CA

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July 10<sup>th</sup>, 2017

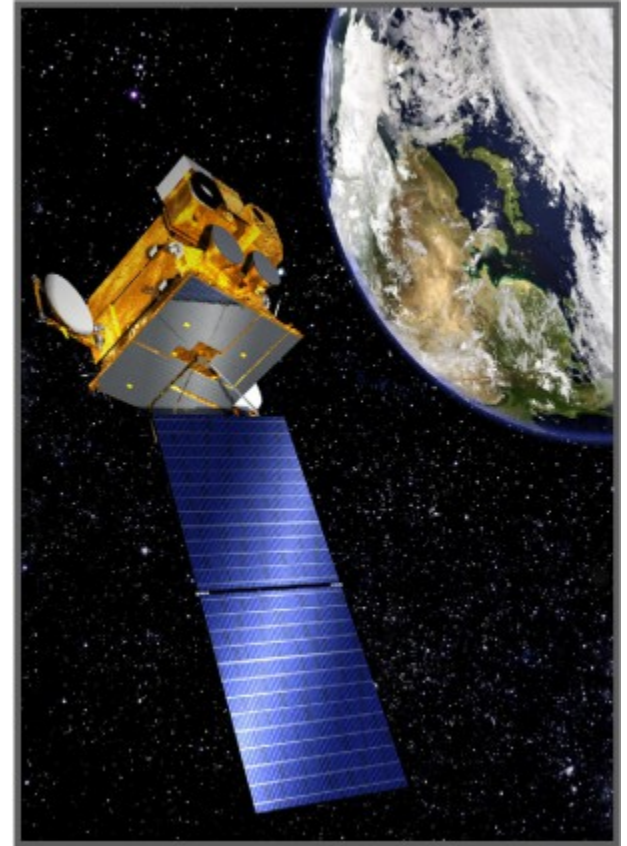
AIAA Propulsion and Energy Forum





# Motivation

- **New chemical propellants are proposed for spacecraft systems**
- **Combustion chemistry is complex**
  - Diagnostics are required to validate models
- **Diagnostics are needed for lifetime analysis**
  - Determine the state-of-health of the system
- **Diagnostics include:**
  - Temperature
  - Species concentration



[1]

# Approach

## Goal:

Develop diagnostic tool for use in microthrusters for measurement of temperature and species concentration

## Challenges:

- Ultra low pressure systems
- Small size
- Caustic chemicals
- Supersonic flow

## Method:

Wavelength modulation spectroscopy  
validated in mock system

- Non-intrusive, in-situ measurements:
  - Temperature
  - Species concentration
- Quantitative and calibration free



# Presentation Outline

---

- 1) Motivation and Approach**
- 2) Introduction to Absorption Spectroscopy**
  - Direct Absorption
  - Wavelength Modulation Spectroscopy
- 3) Testing**
  - Optical system
  - Experimental setup
  - Scanning path
- 4) Results**
  - Data processing
  - Absorbance area
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    - Path averaged
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  - Species Concentration
- 5) Conclusions and Acknowledgements**

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# Absorption Spectroscopy

- Molecules absorb light at discrete wavelengths resonant with molecular transitions

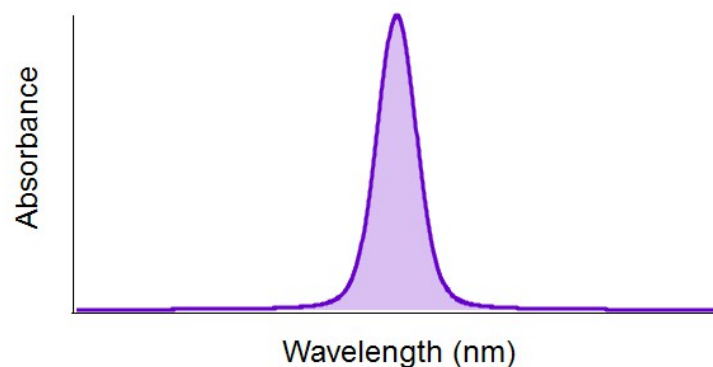
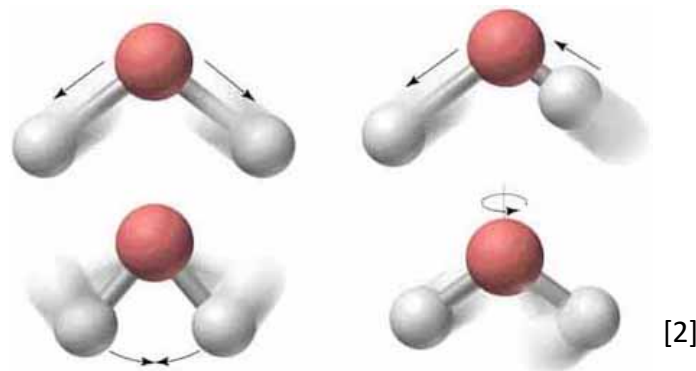
- The amount of light absorbed is related to:

$$\alpha(\lambda) = S(T) P X L \Phi$$

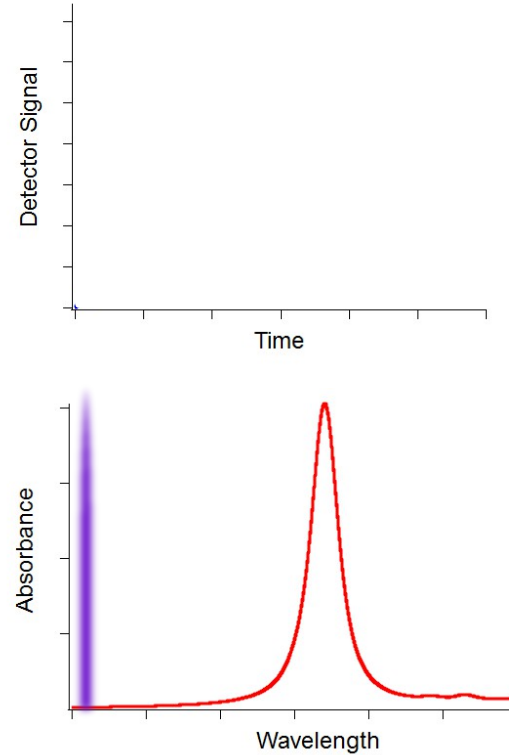
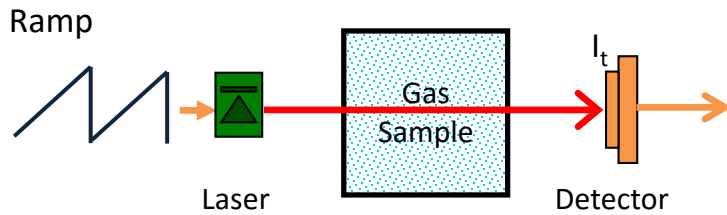
- Line-strength:  $S(T)$
- Pressure:  $P$
- Mole fraction:  $X$
- Pathlength:  $L$
- Line-shape function:  $\phi$

- Parameters:  $S(T)$  and  $\phi$  are unique to each transition

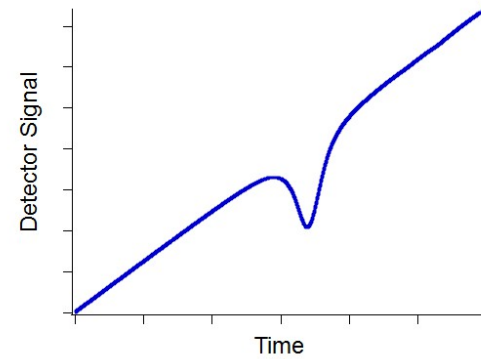
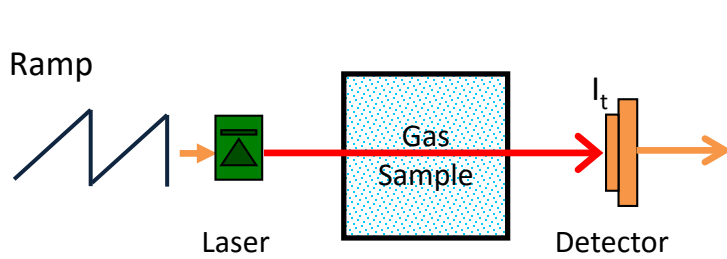
- HITRAN2012 [3]
- Experimental validations [4,5]



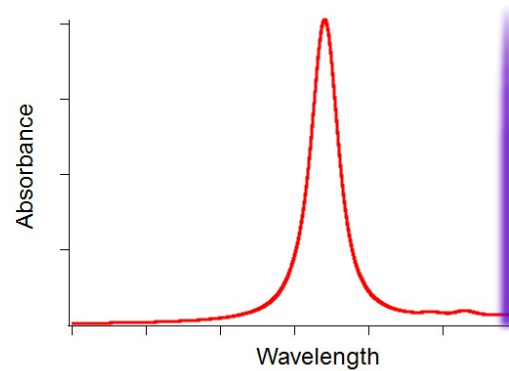
# Tunable Diode Laser Absorption Spectroscopy (TDLAS)



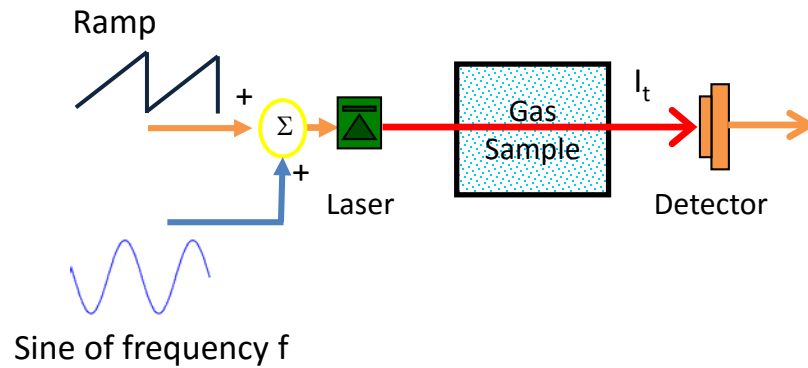
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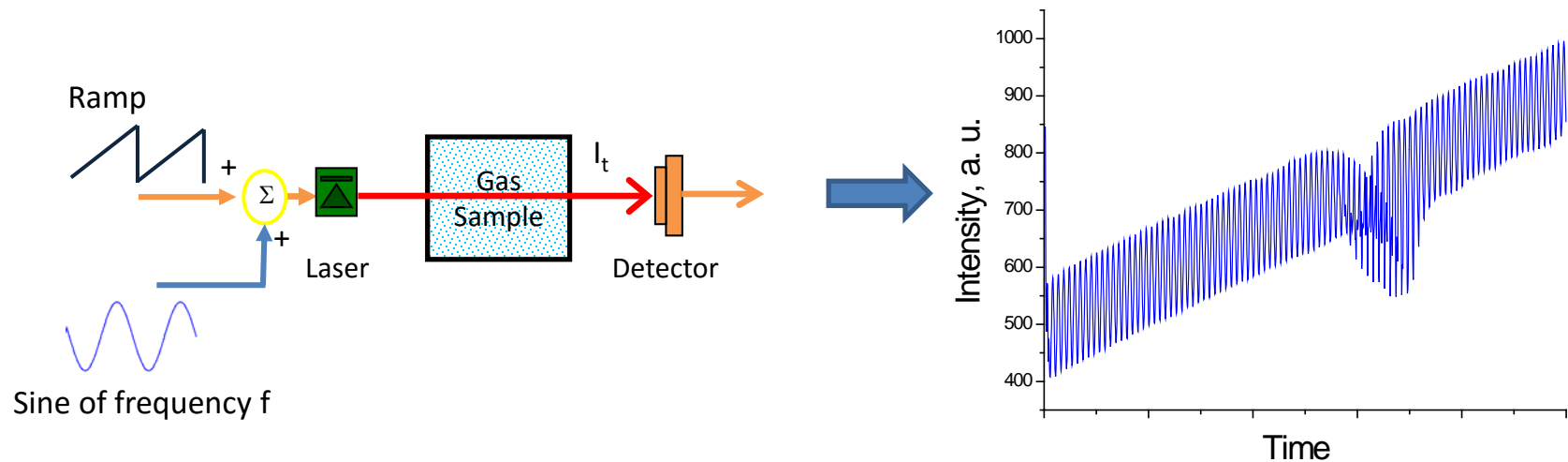
$$\alpha(\lambda) = S(T)PL\Phi = -\ln\left(\frac{I}{I_o}\right)$$



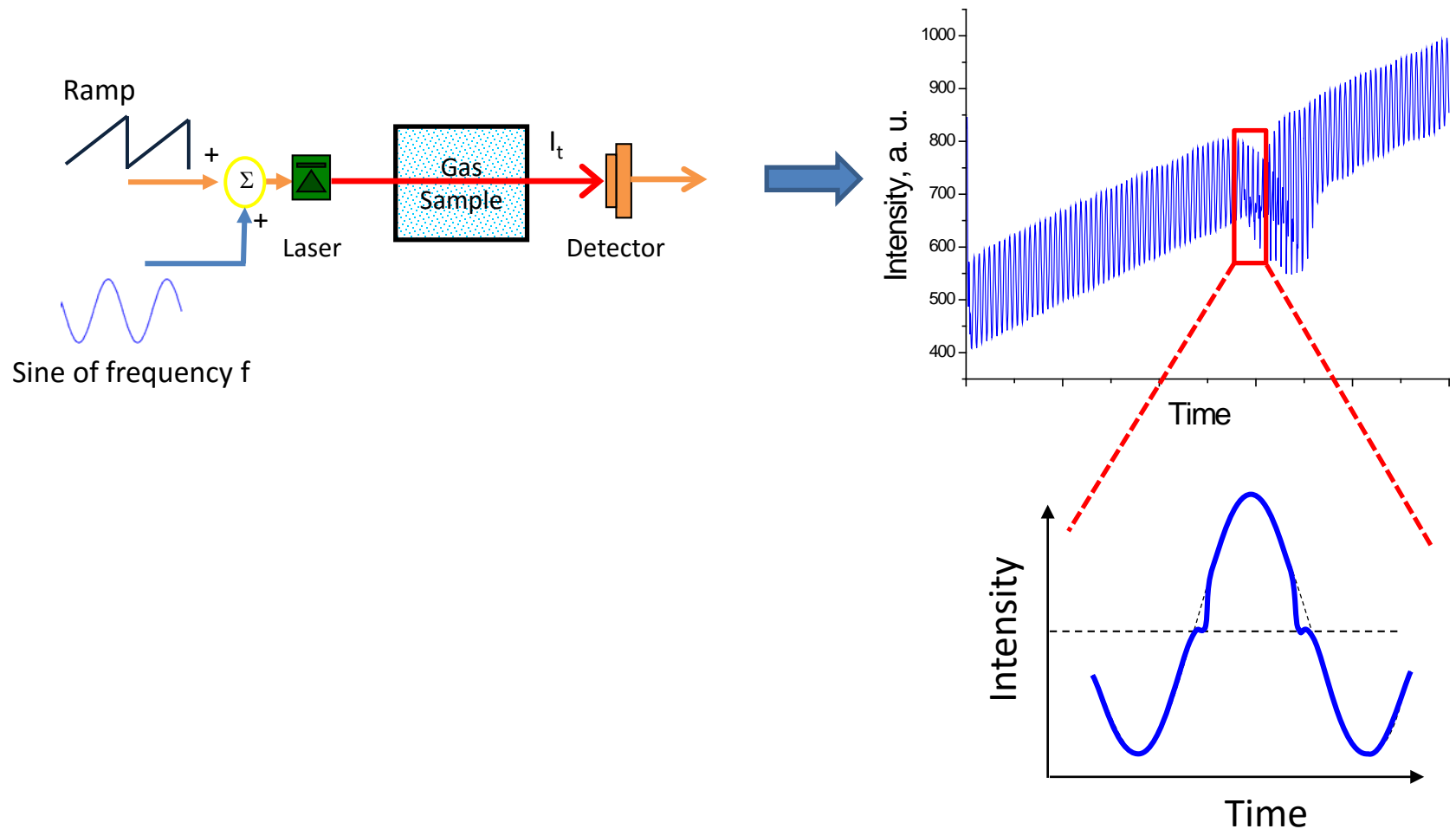
# Wavelength Modulation Spectroscopy



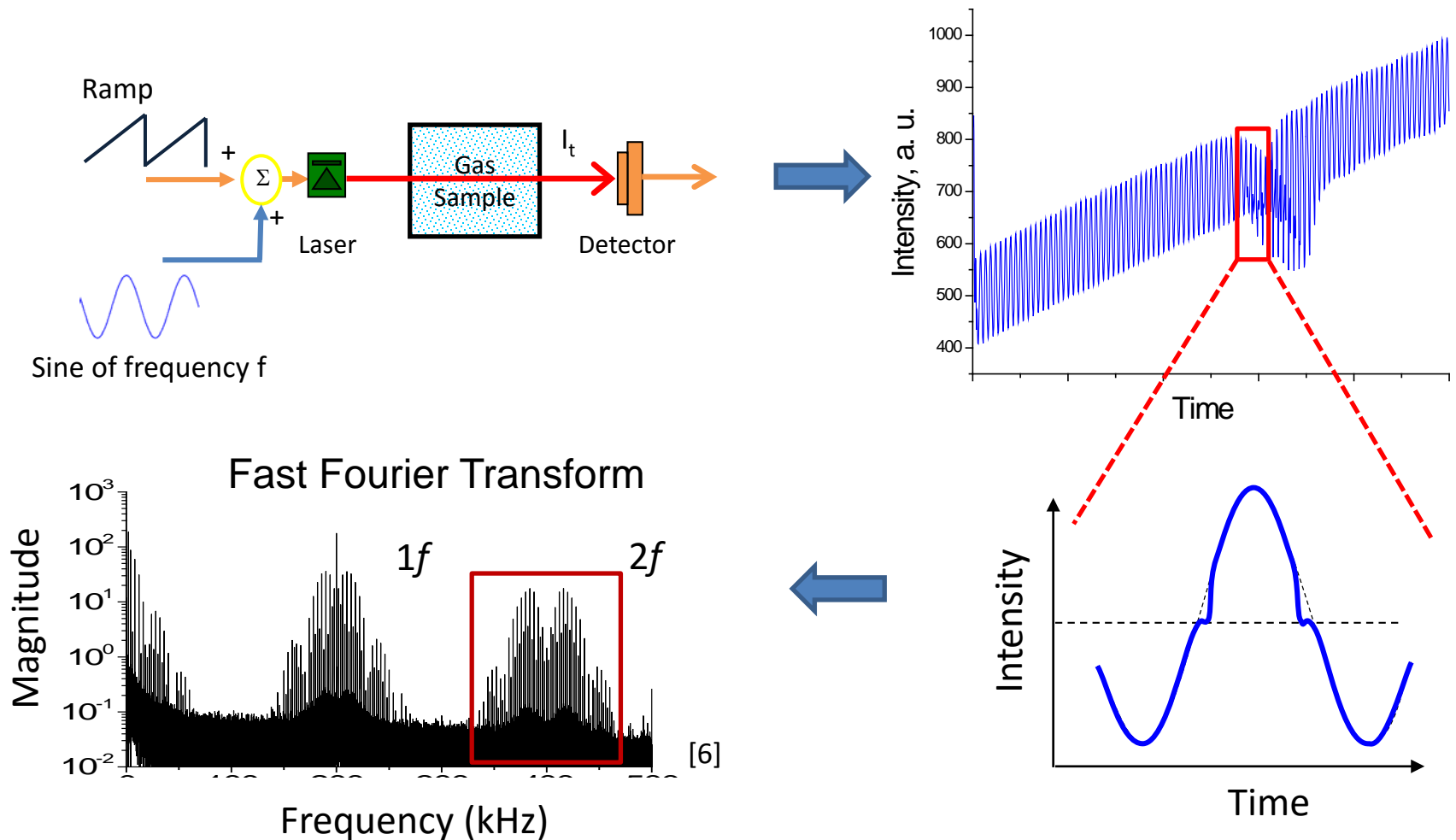
# Wavelength Modulation Spectroscopy



# Wavelength Modulation Spectroscopy

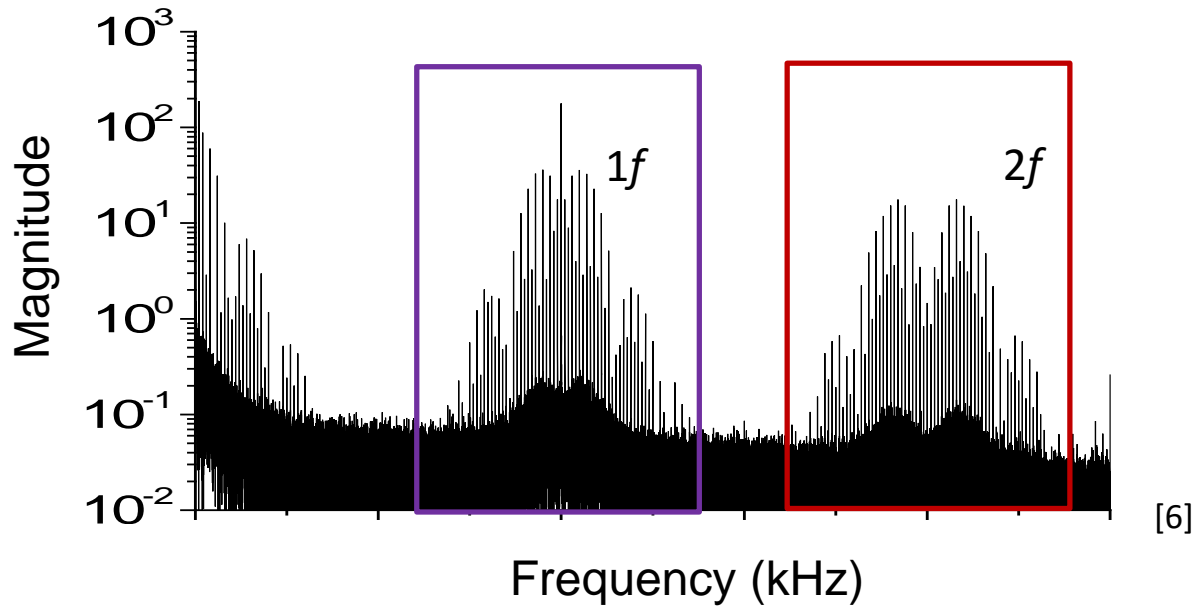


# Wavelength Modulation Spectroscopy

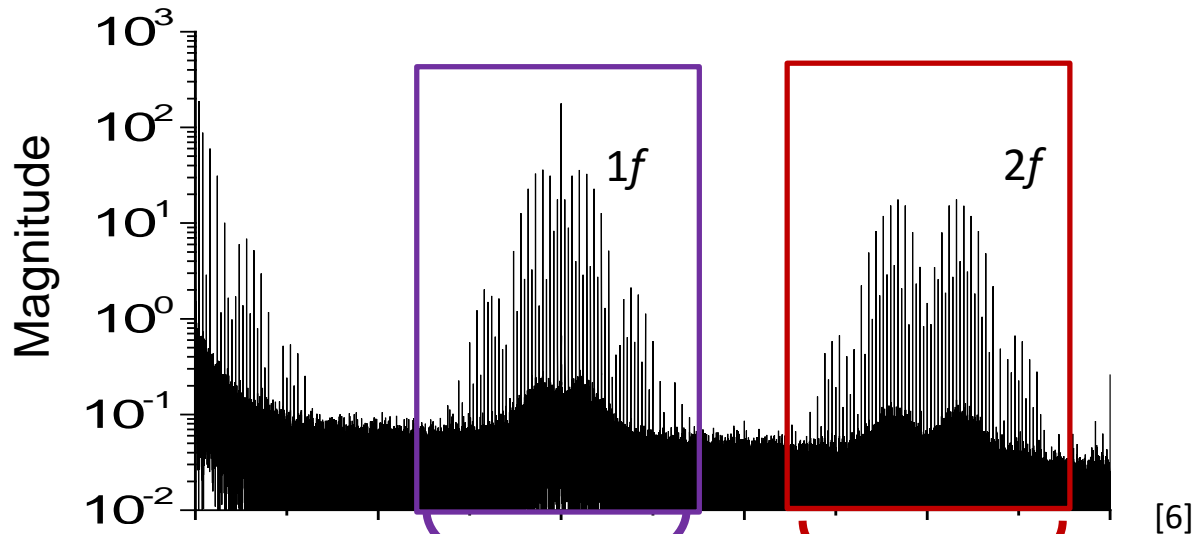




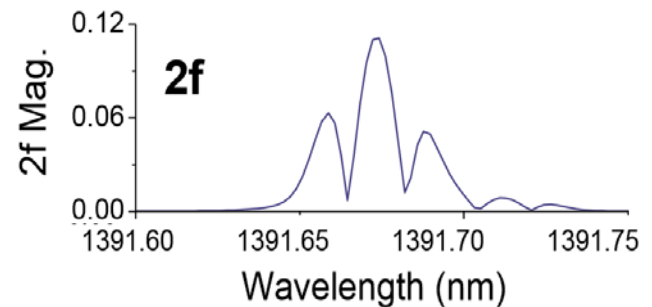
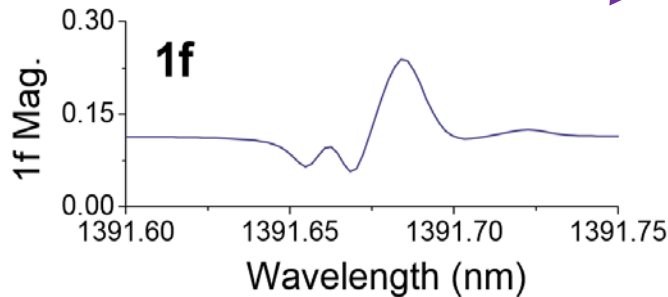
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# Wavelength Modulation Spectroscopy

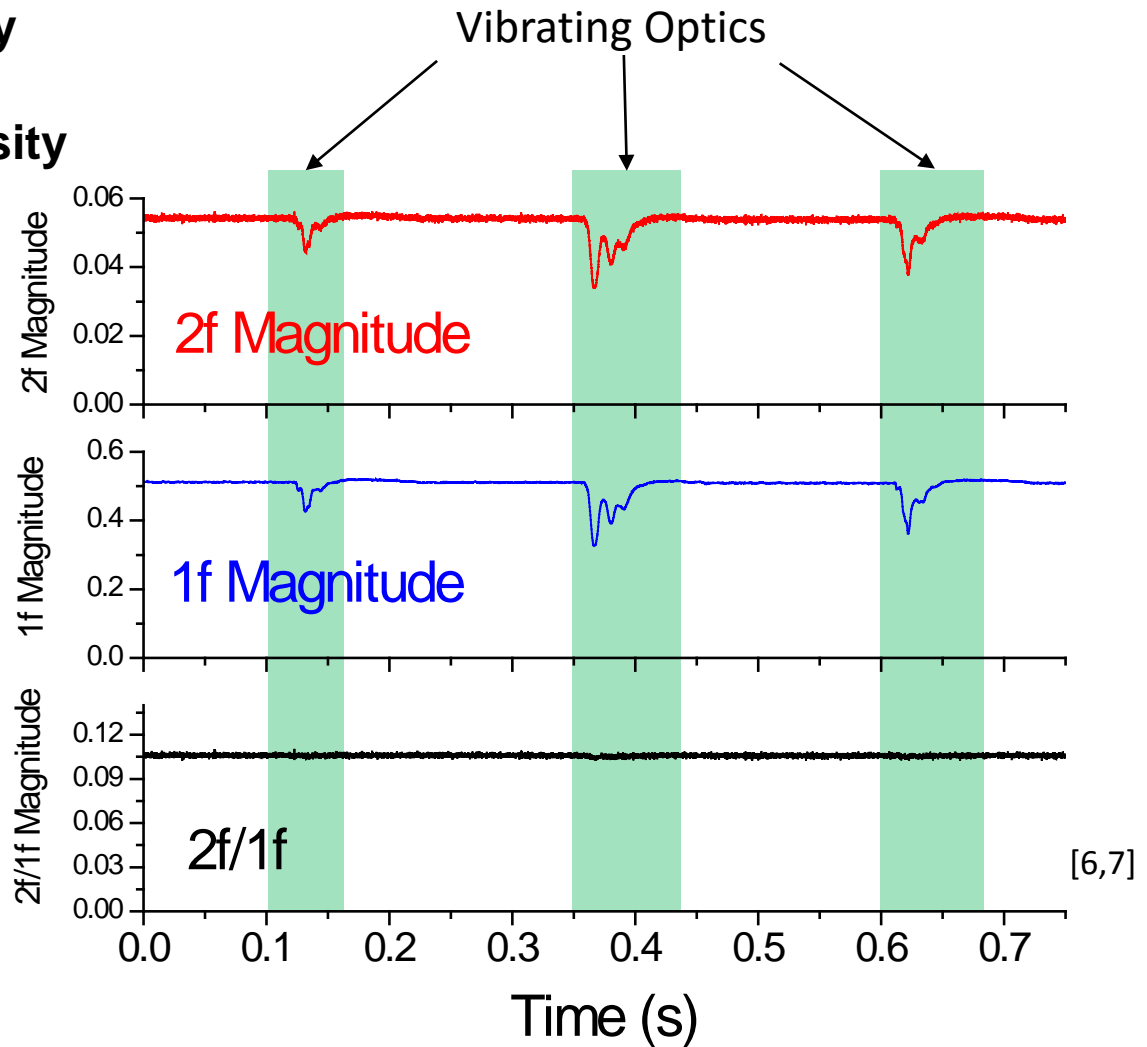
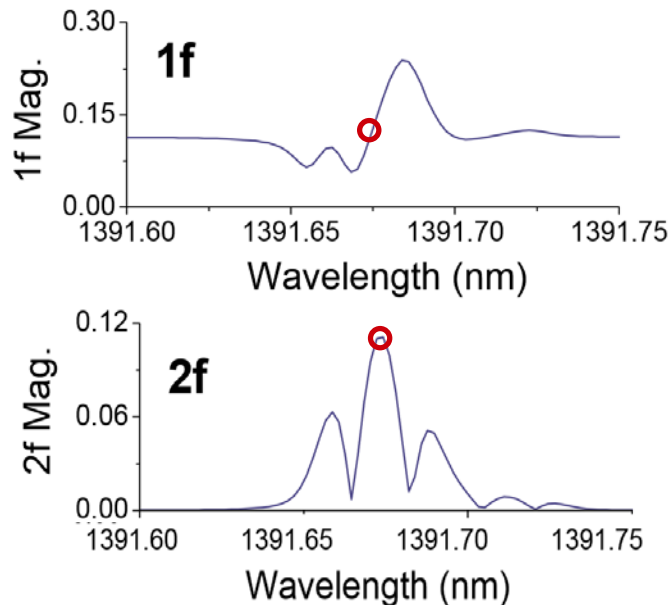


Apply lockin amplifier

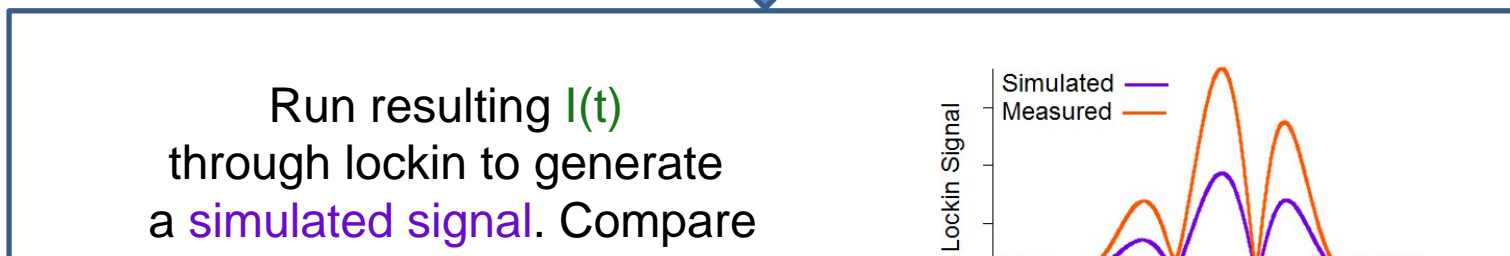
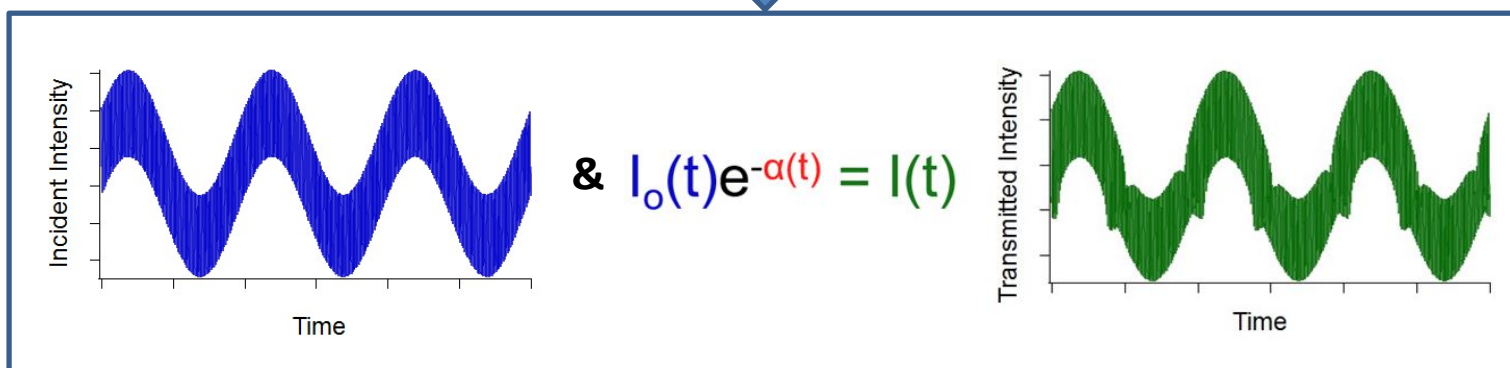
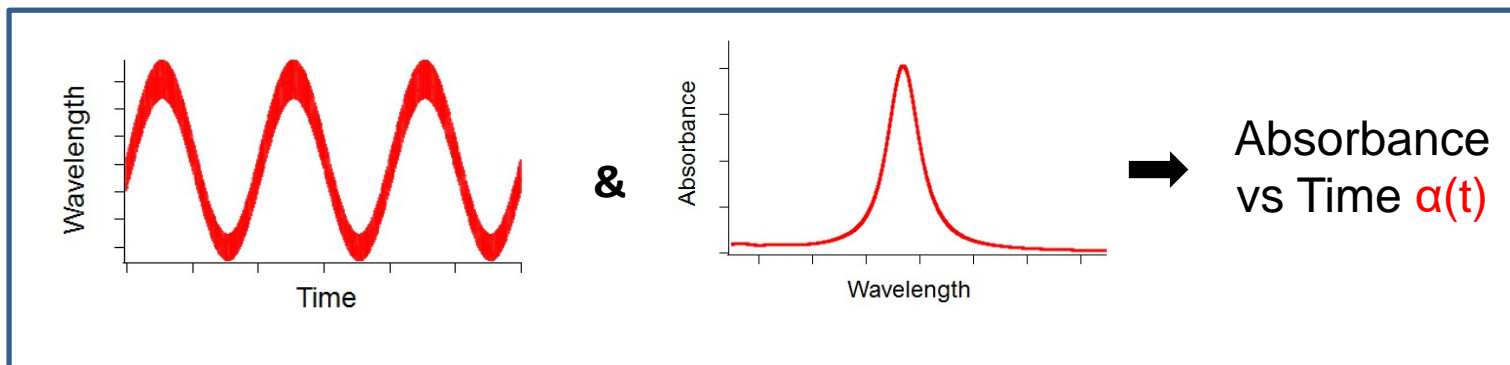


# Wavelength Modulation Spectroscopy

- Normalizing the 2f signal by the 1f signal removes dependence on laser intensity
- Important in harsh environments
  - Window fouling
  - Vibrating optics



# Data Processing



# Temperature Measurement

Temperature can be calculated from the ratio of the absorbance areas of two well chosen absorption features

$$\text{Absorbance area ratio} = \frac{S1(T)PXL}{S2(T)PXL}$$

[10]

# Temperature Measurement

Temperature can be calculated from the ratio of the absorbance areas of two well chosen absorption features

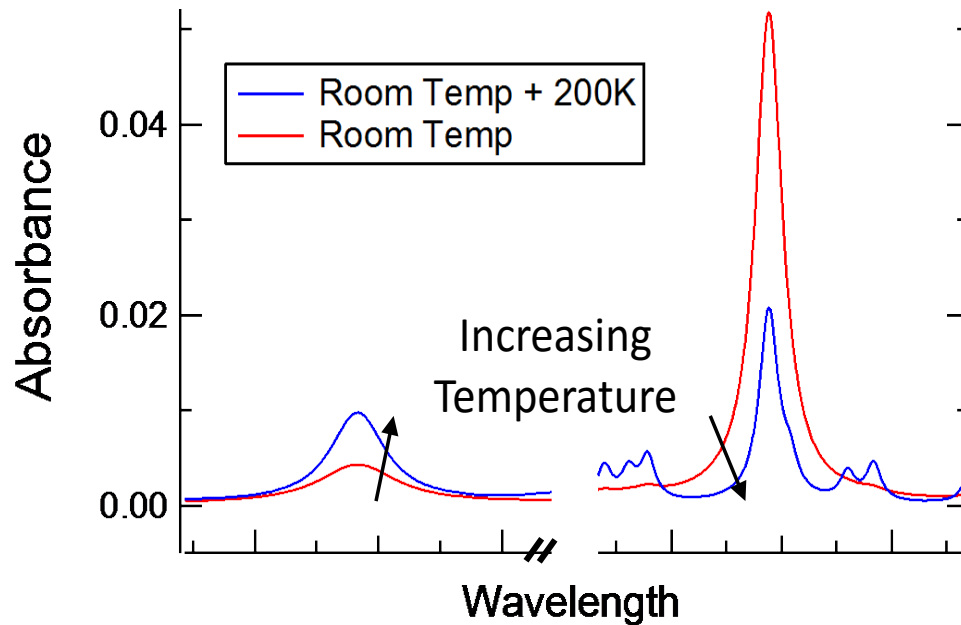
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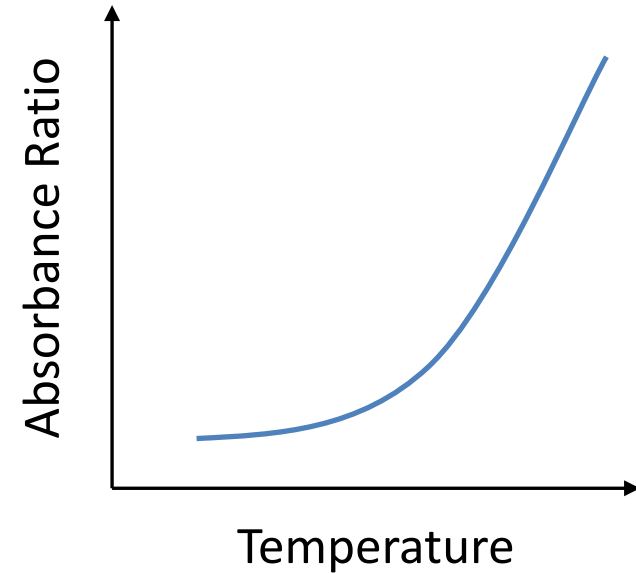
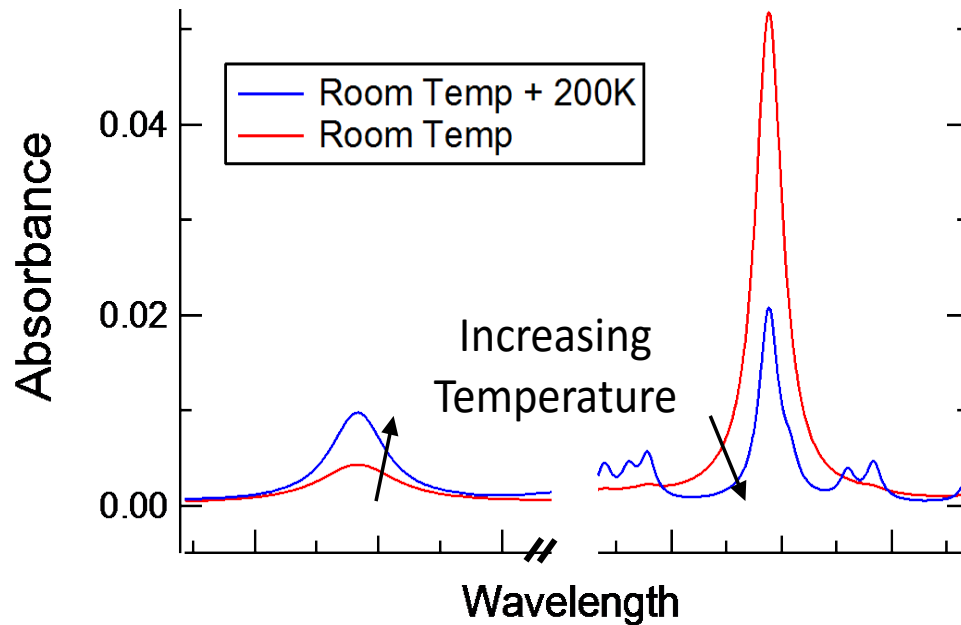
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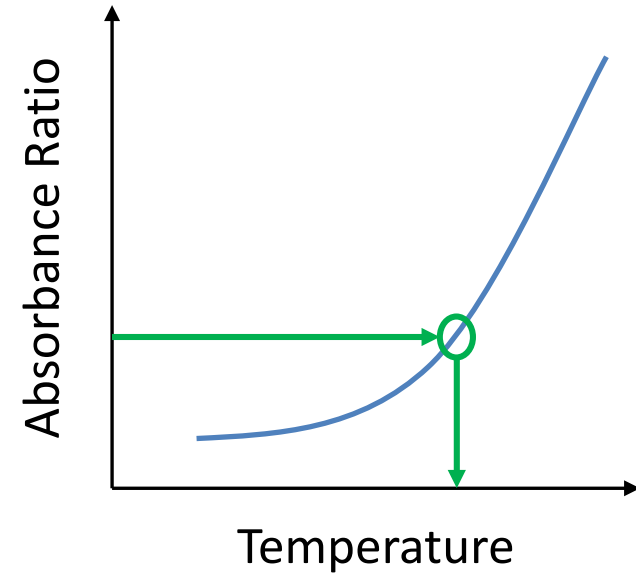
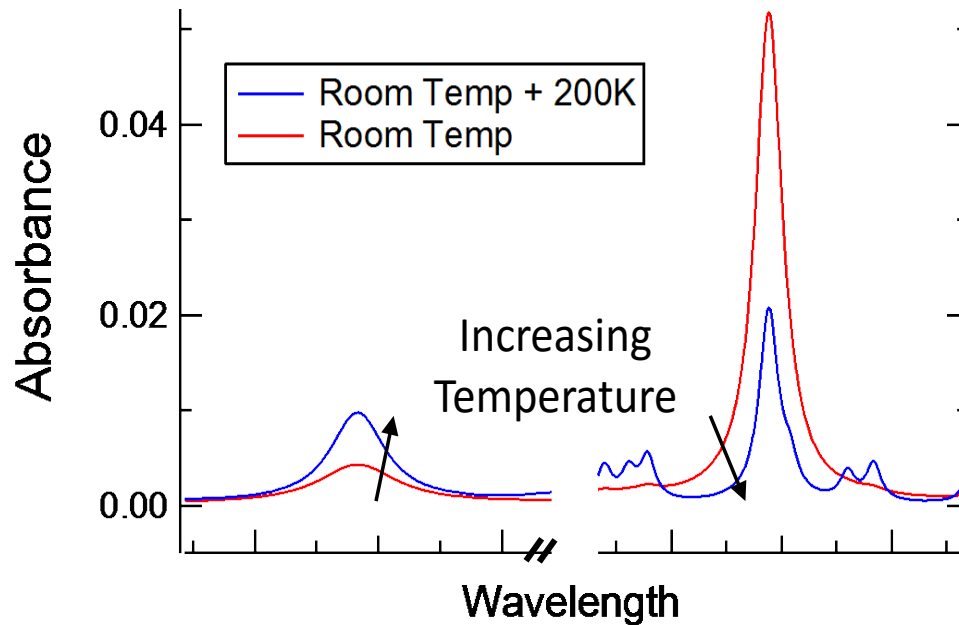


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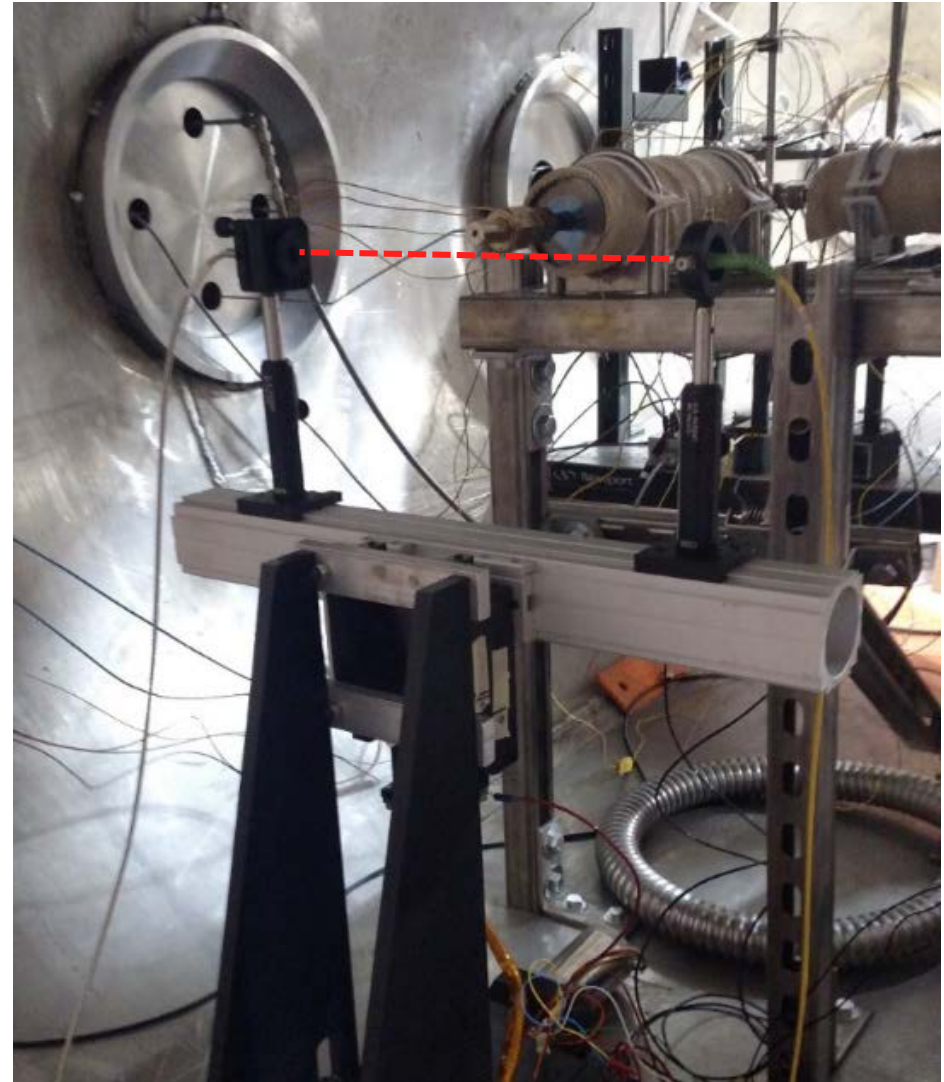
[10]

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    - Abel inversion
  - Species Concentration
- 5) **Conclusions and Acknowledgements**

# System Design

- **Testing performed at the AFRL at Edwards Air Force Base**
  - Vacuum chamber 4 within the Spacecraft Propulsion Laboratory
- **System developed by the AFRL**
  - Exhaust products:  $\text{NH}_3$  and  $\text{H}_2\text{O}$
  - Expansion ratio: 5.19
  - Exit Mach number: 3.1
- **LabView controlled translation stage mounted downstream**



# Spectroscopy System

- **Spectroscopy system developed at CU Boulder**

- Verification of  $\text{NH}_3$  and  $\text{H}_2\text{O}$  line parameters in furnace
- Characterized laser operating conditions
- Developed simulation procedures

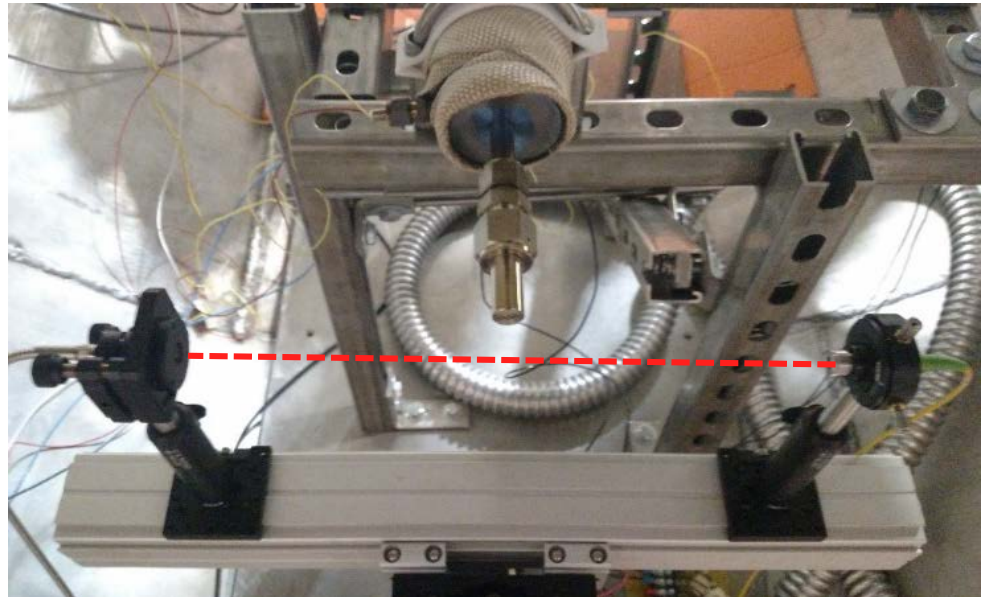


- **Measurements**

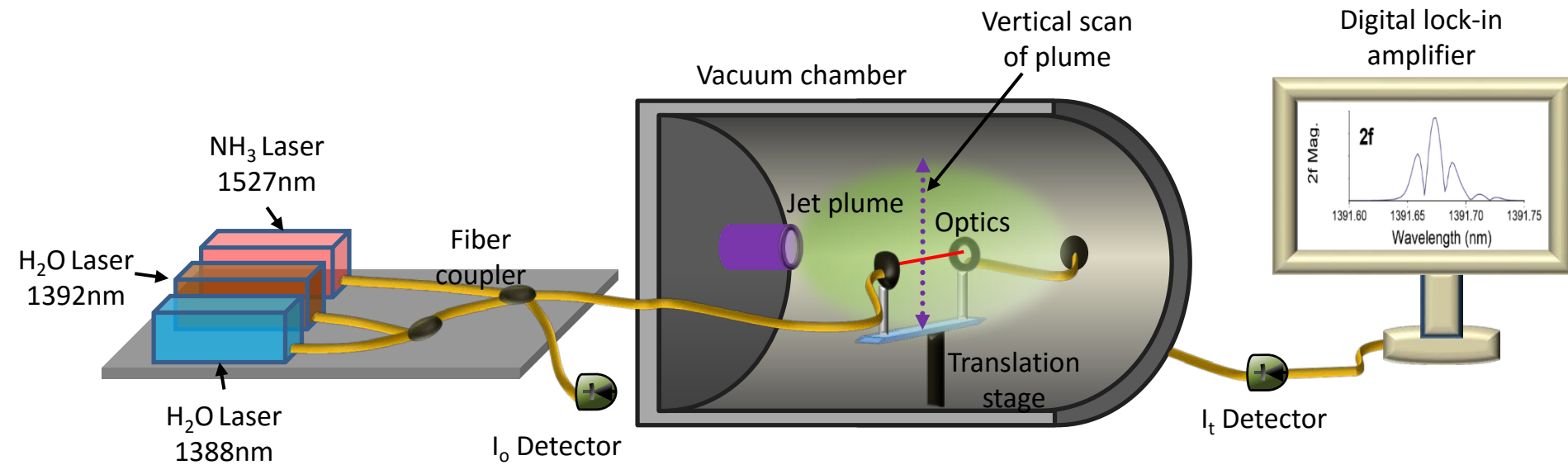
- Temperature distributions
- $\text{H}_2\text{O}$  concentration
- $\text{NH}_3$  concentration

- **Three Near-IR lasers**

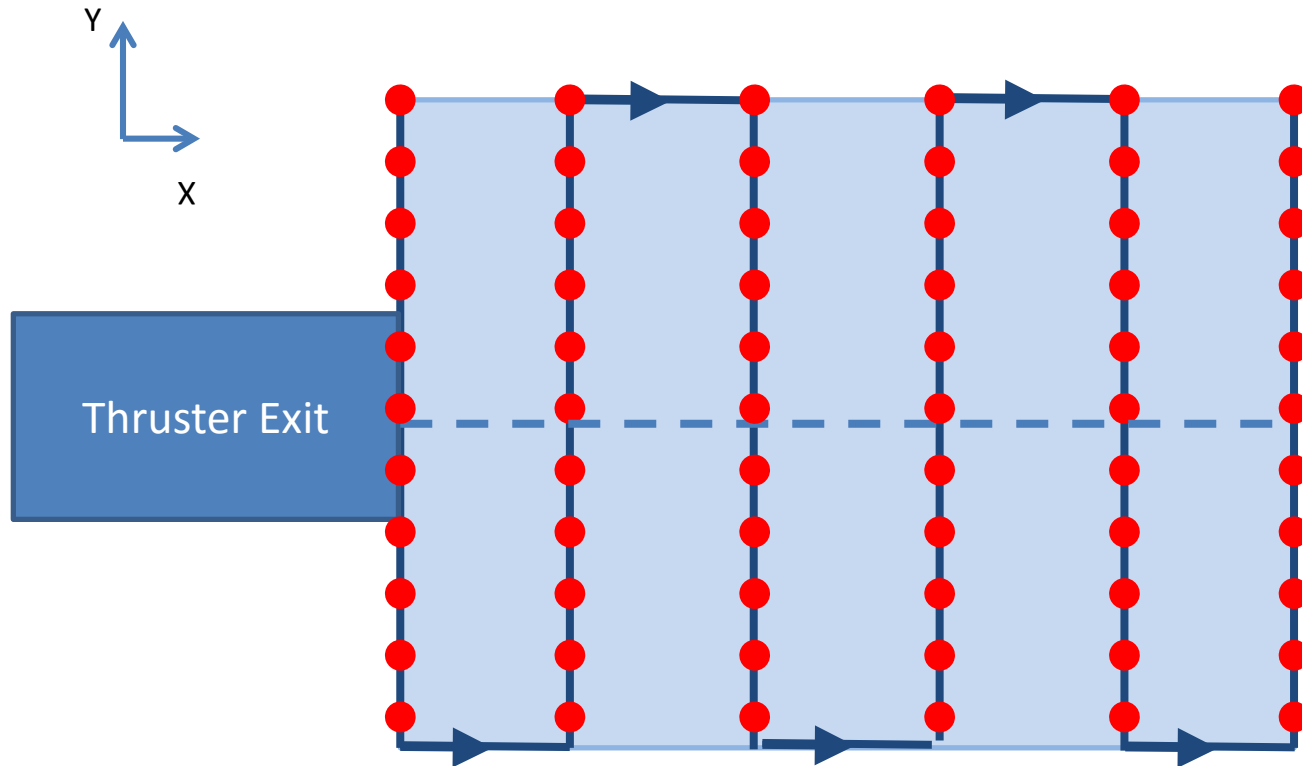
- 1388nm ( $\text{H}_2\text{O}$ )
- 1392nm ( $\text{H}_2\text{O}$ )
- 1527nm ( $\text{NH}_3$ )



# Experimental Setup



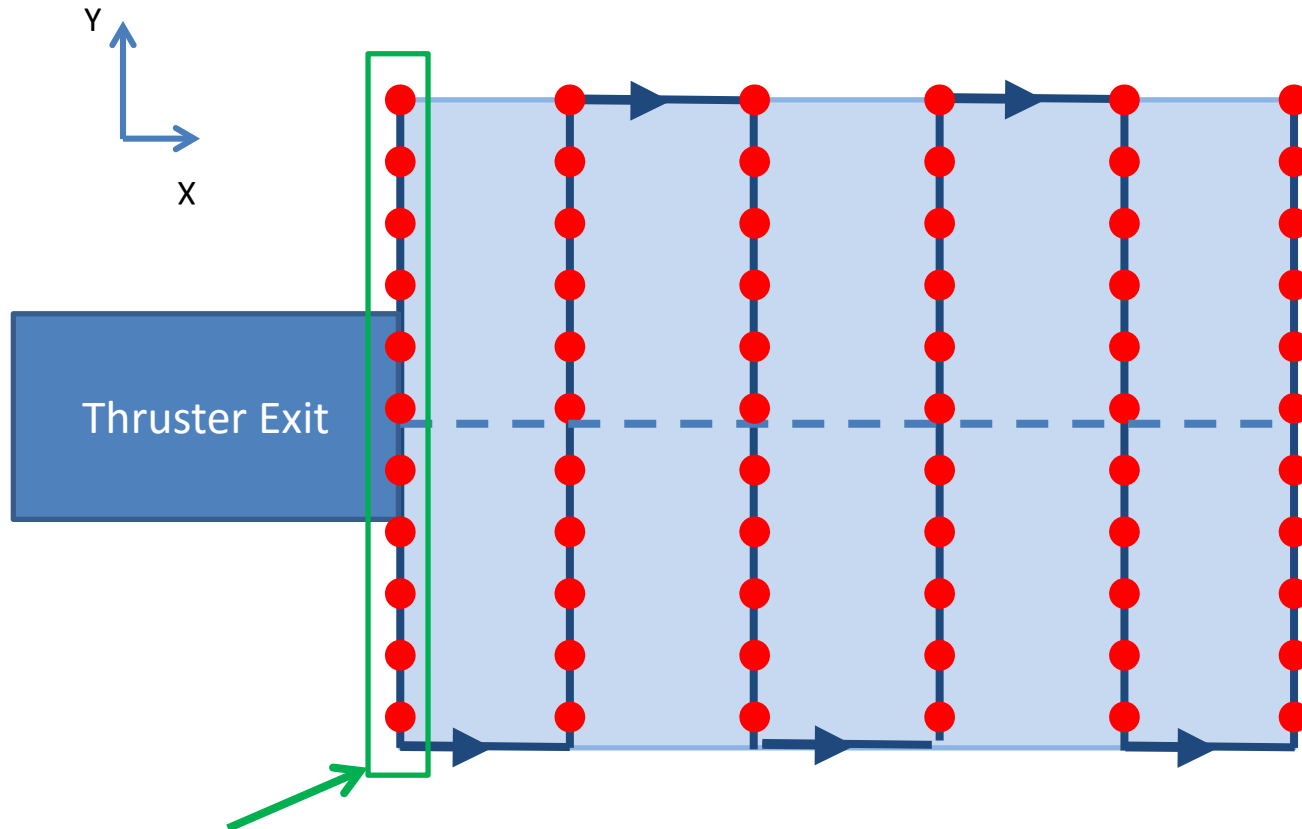
# Scanning Path



$x = 1 \text{ mm}$  (from exit plane of nozzle), 81 point vertical scan,  $\Delta y = 0.1 \text{ mm}$



# Scanning Path



$x = 1 \text{ mm}$  (from exit plane of nozzle), 81 point vertical scan,  $\Delta y = 0.1 \text{ mm}$

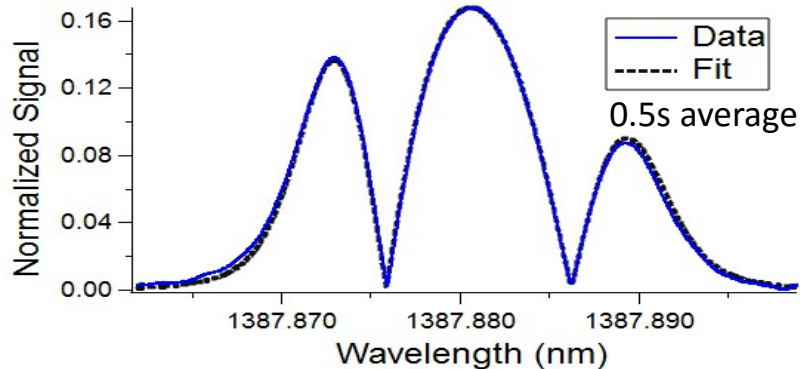
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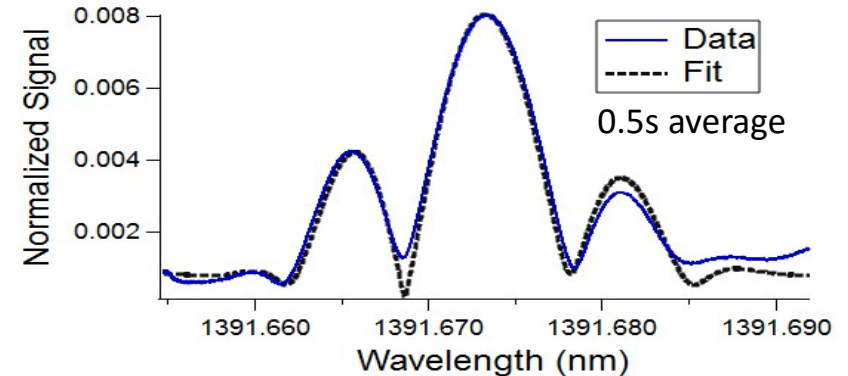
# Data Processing

## Water absorption features

Typical fit for 1388nm signal ( $\text{H}_2\text{O}$ )

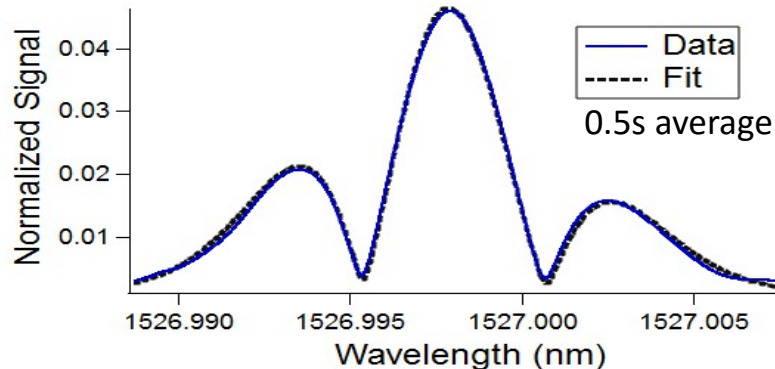


Typical fit for 1392nm signal ( $\text{H}_2\text{O}$ )



## $\text{NH}_3$ absorption feature

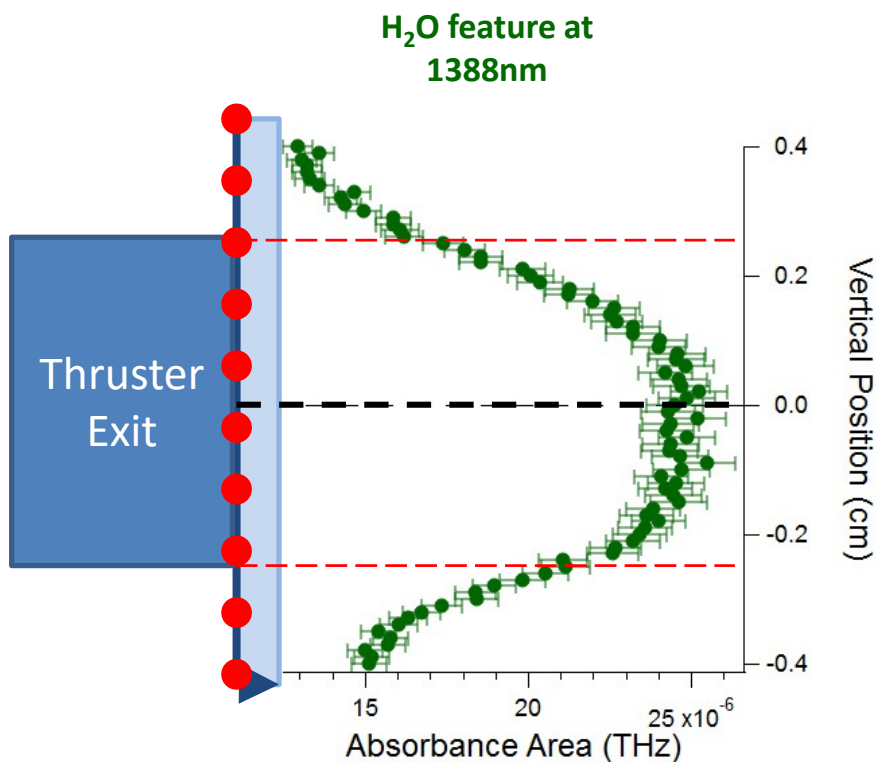
Typical fit for 1527nm signal ( $\text{NH}_3$ )



**Good agreement between the simulated signal and the fit indicate correct absorbance parameters**

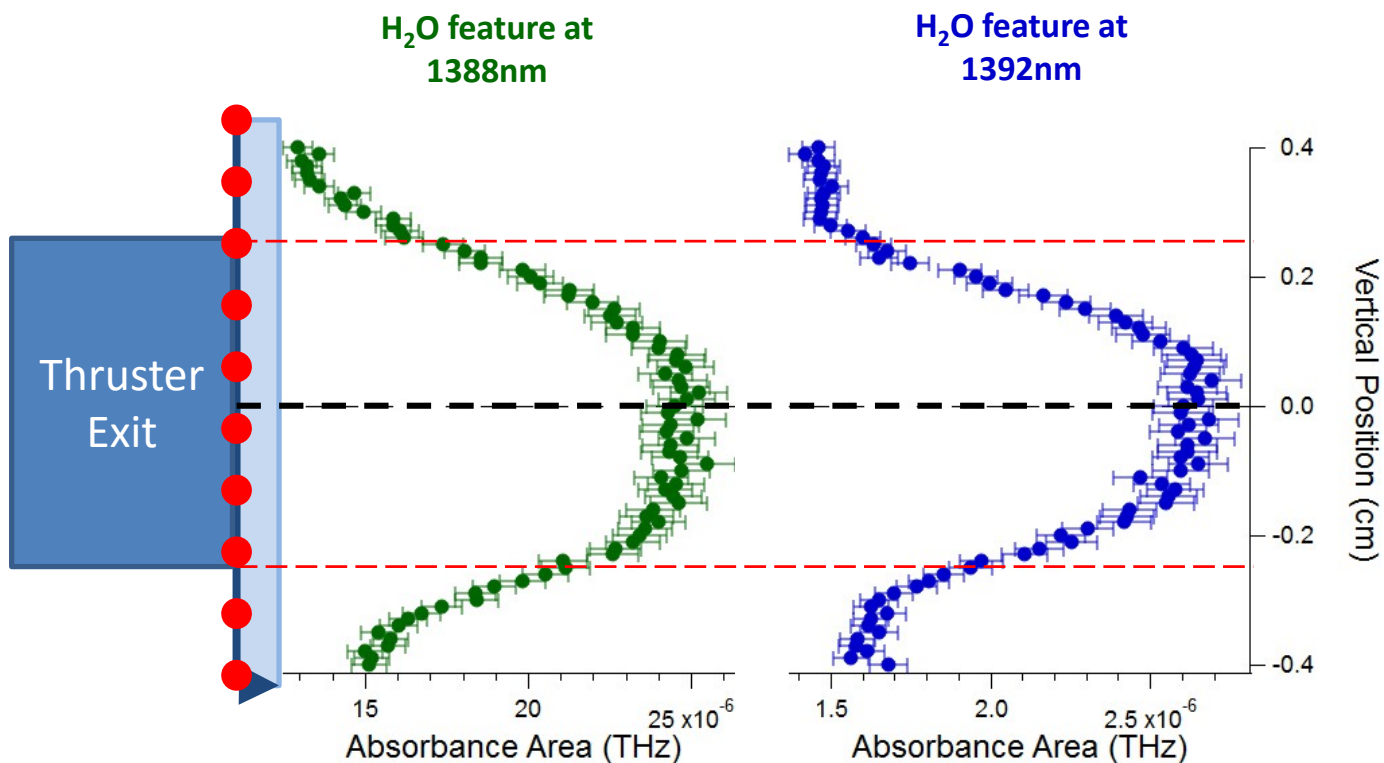
[8,9]

# Absorbance Area



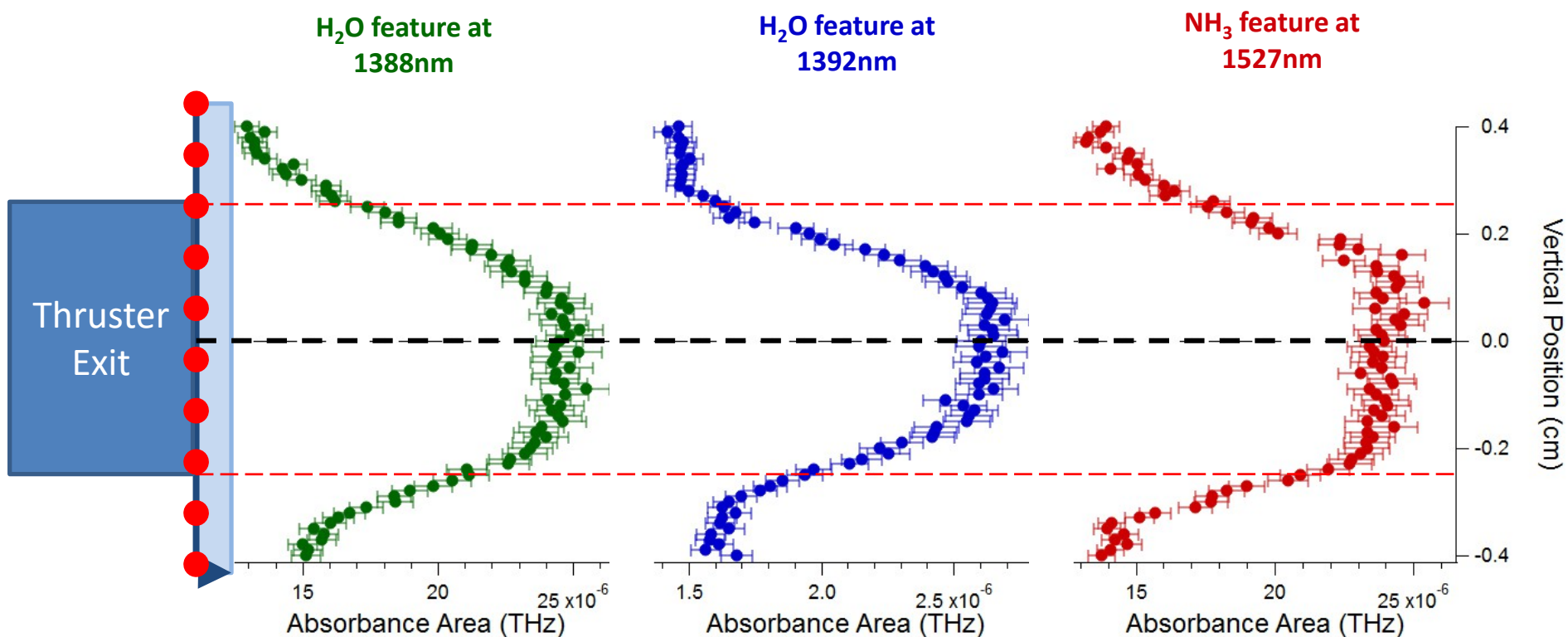
Data recorded at 1mm from exit plane of thruster

# Absorbance Area



Data recorded at 1mm from exit plane of thruster

# Absorbance Area

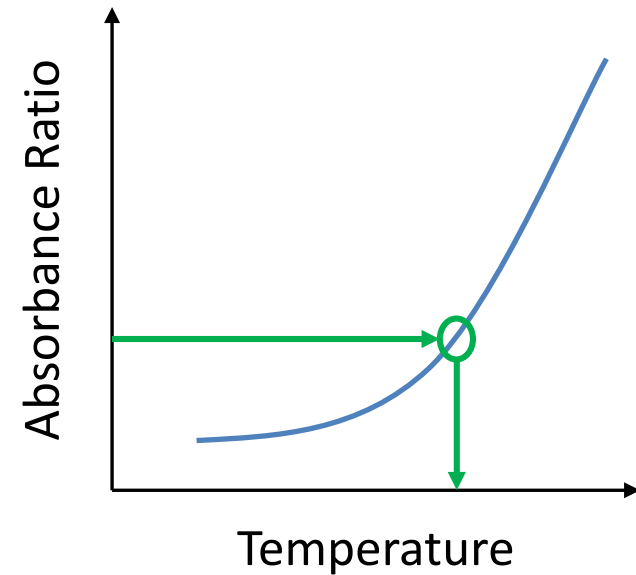
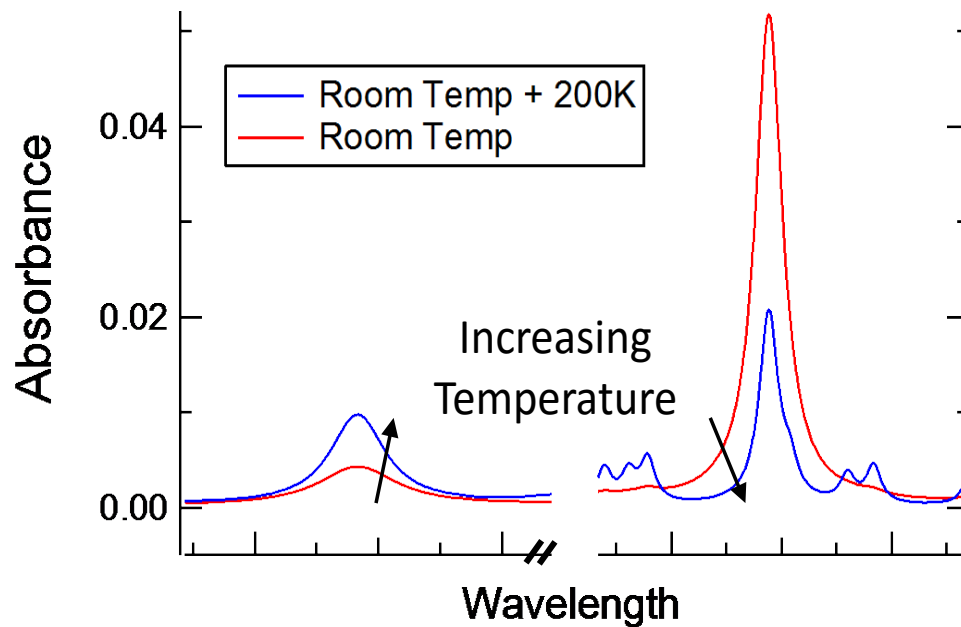


Data recorded at 1mm from exit plane of thruster

# Temperature Measurement

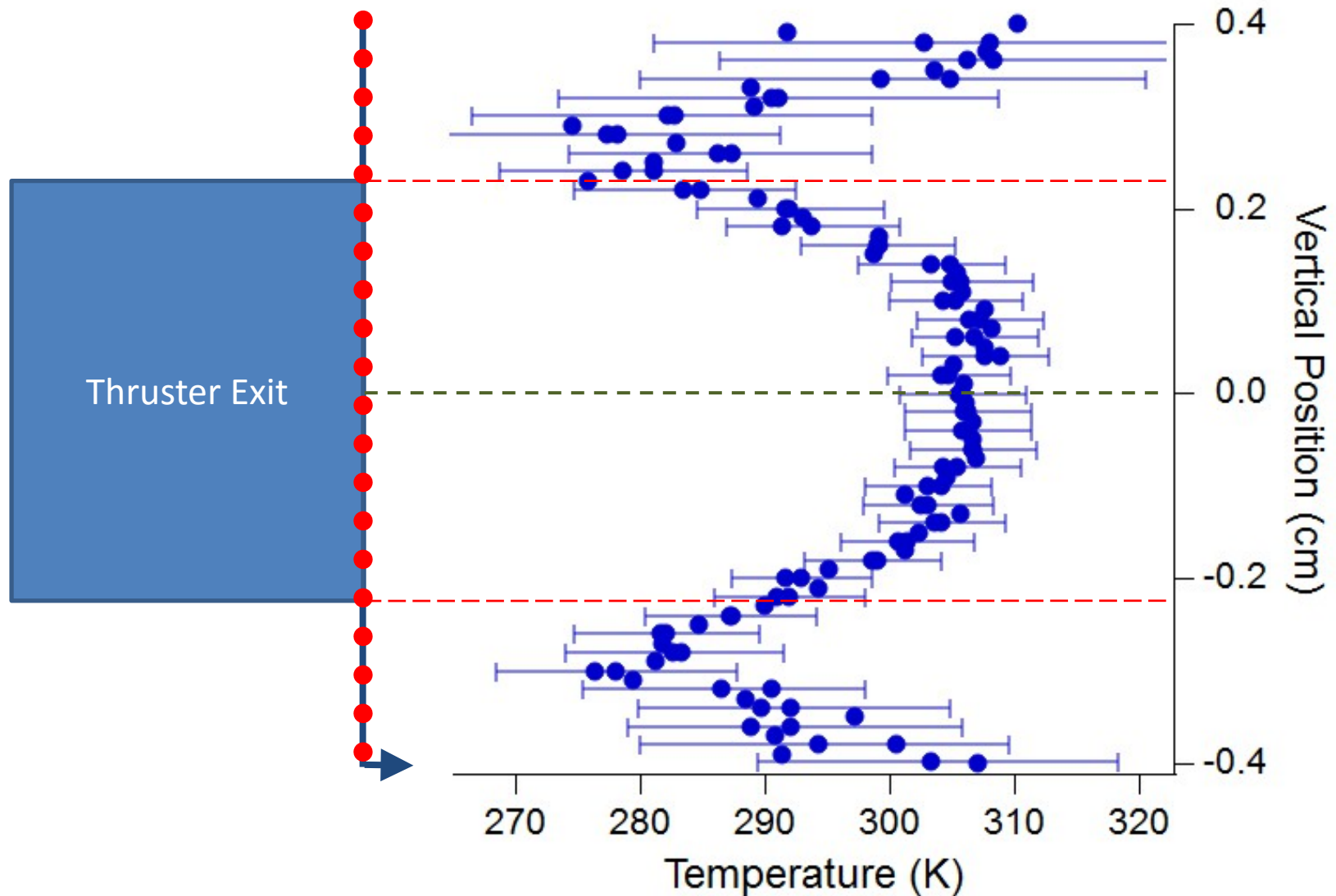
Temperature can be calculated from the ratio of the areas for 1388nm and 1392nm absorption features

$$\text{Absorbance area ratio} = \frac{S(1388) \cancel{PXL}}{S(1392) \cancel{PXL}} = \frac{S(1388)}{S(1392)}$$



[10]

# Temperature Measurement



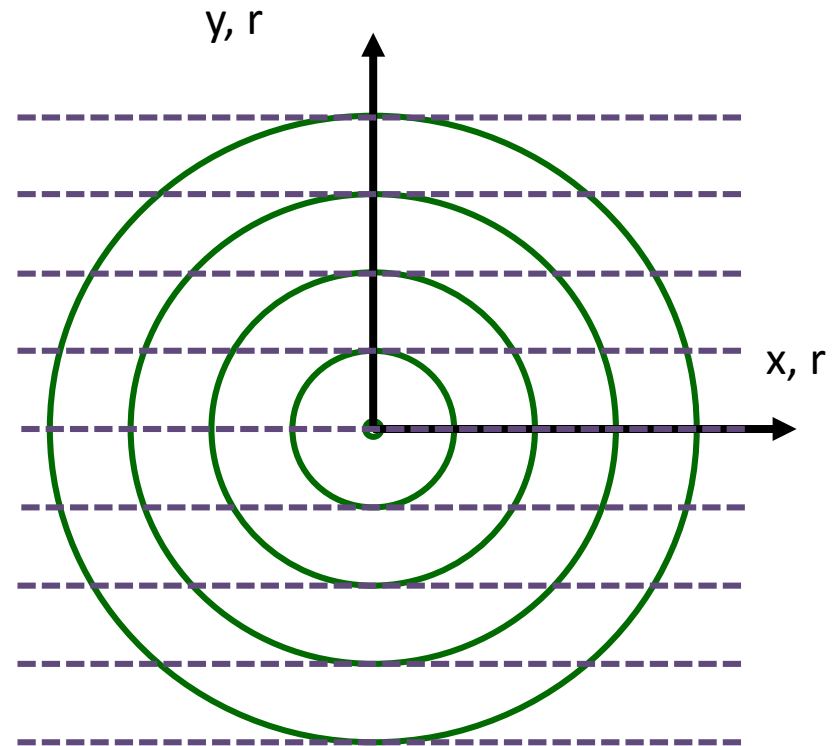


# Inverse Abel Transform

- **Apply inverse Abel transform to path averaged measurements**

- Assuming axisymmetric flow
- Calculate radial distributions
- Use “Fourier Method” to approximate  $F(y)$  as a cosine expansion

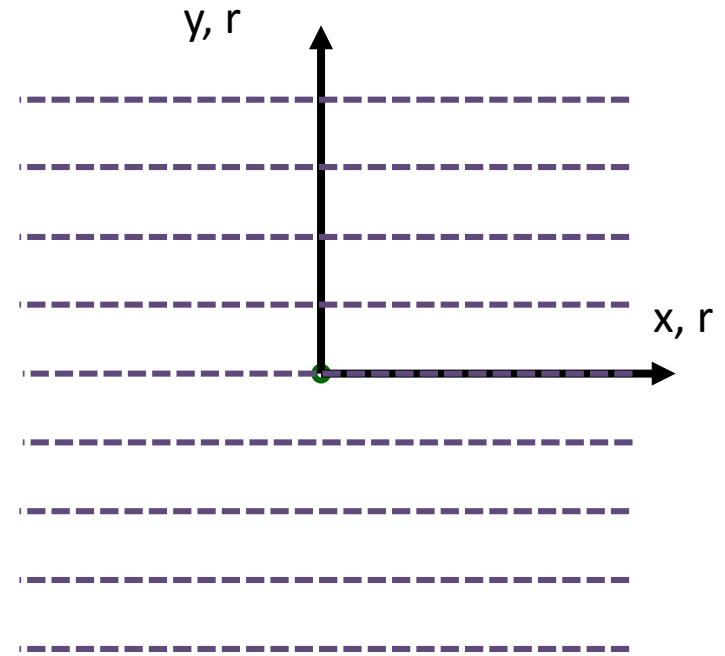
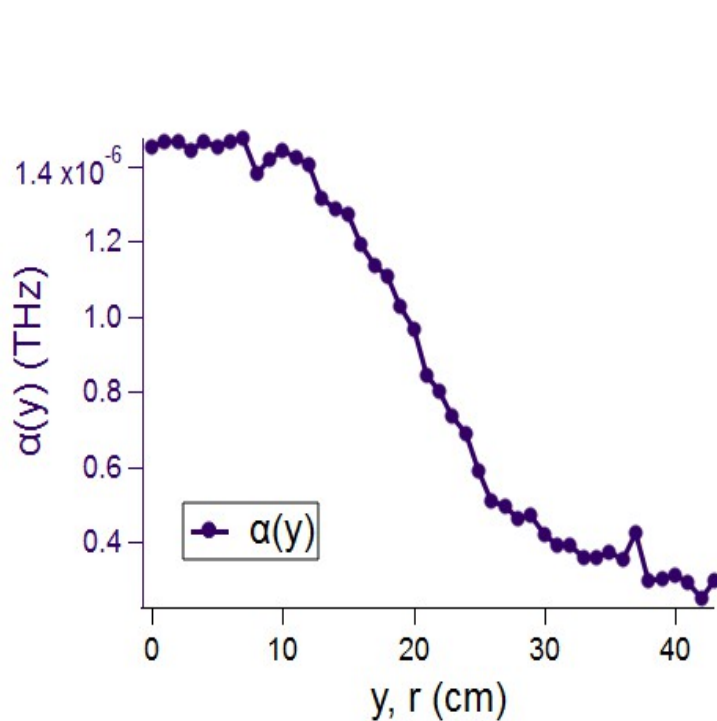
$$f(r) = -\frac{1}{\pi} \int_y^R \frac{dF(y)}{dy} \frac{dy}{\sqrt{y^2 - r^2}}$$



[11,12]

# Inverse Abel Transform

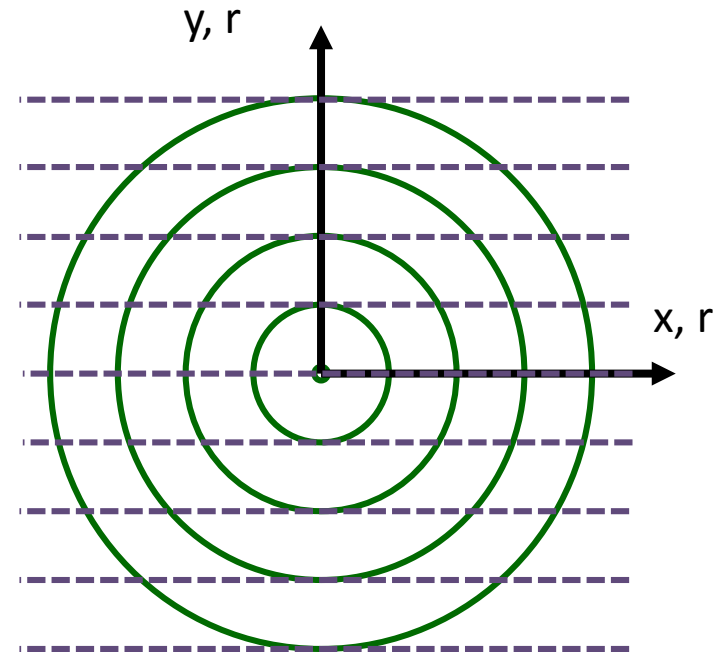
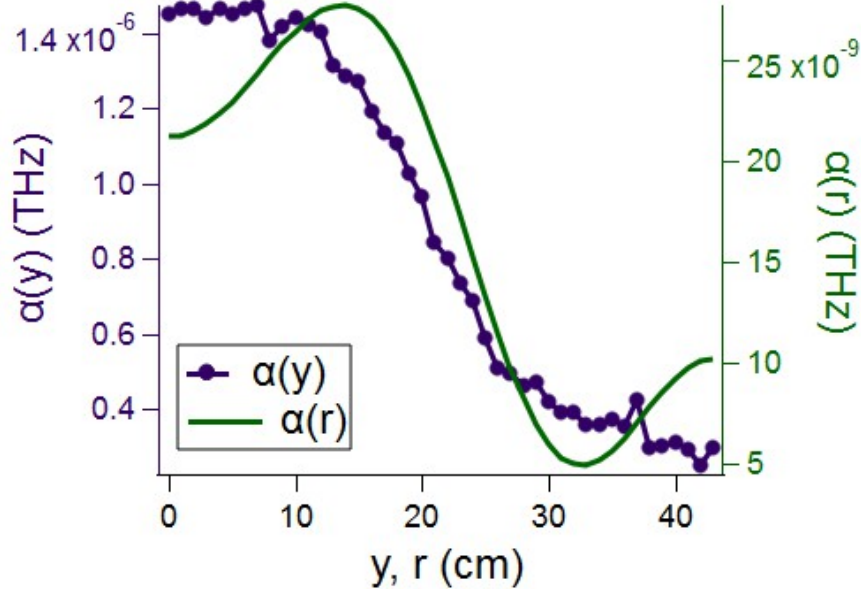
## Absorbance Area of 1388 Feature



[11,12]

# Inverse Abel Transform

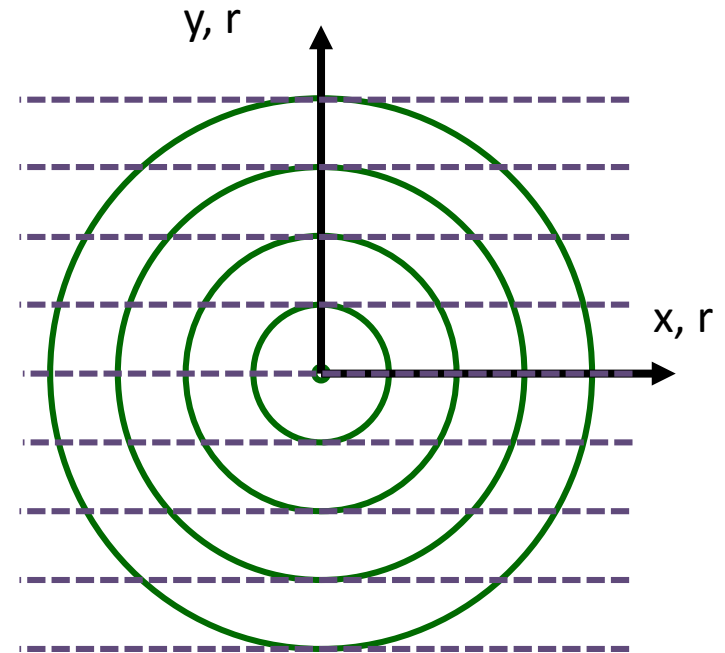
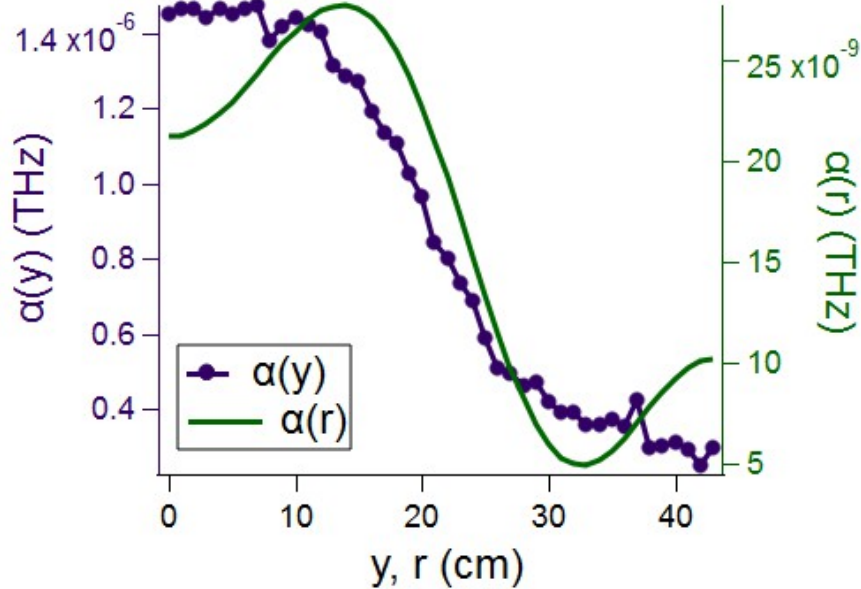
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[11,12]

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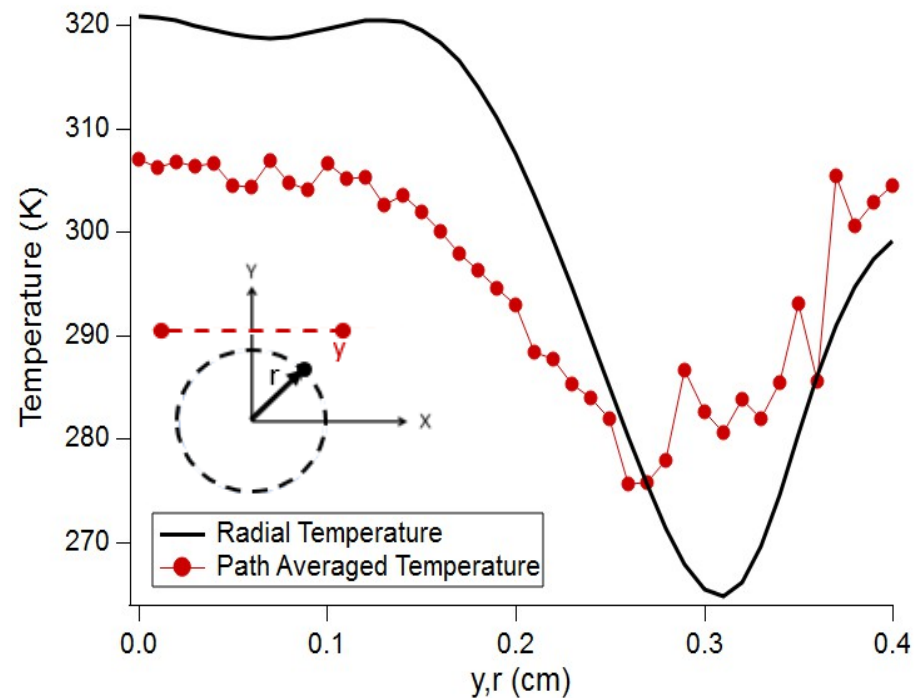
## Absorbance Area of 1388 Feature



[11,12]

# Inverse Abel Transform

- **Isentropic calculations predict core temperature of 353K at outlet**
  - Results are 1mm downstream of outlet
  - Magnitude of temperature drop agrees with Naik et al. (2009) and Woodmansee et al. (2004)
- **Transverse temperature trends are consistent with Woodmansee et al. (2004)**
  - Exit plane temperature measurements
- **Lower temperature regions coincide with edges of nozzle**



[11,12]

# Species Concentration

- **Traditionally to calculate concentration (X) from absorbance area we must know pressure and path length**
  - Pressure distribution is unknown at measurement location
  - Edges of the plume are not distinctly defined
- **Computational models could be used**
  - Time intensive
  - Could introduce large errors if not correct
- **Simulating pressure distributions adds complexity to absorbance area simulation**

$$\alpha(\lambda) = S(T)PX L\Phi$$

# Species Concentration

- **New technique similar to the temperature measurement technique shown earlier**

- Take the ratio of absorbance areas of the  $\text{NH}_3$  feature and the 1388 nm  $\text{H}_2\text{O}$  feature

$$\frac{A_{1388}}{A_{1527}} = \frac{S_{1388}(T)X_{\text{H}_2\text{O}}\cancel{LP_{\text{total}}}}{S_{1527}(T)X_{\text{NH}_3}\cancel{LP_{\text{total}}}} = \frac{S_{1388}(T)X_{\text{H}_2\text{O}}}{S_{1527}(T)X_{\text{NH}_3}} \quad (1)$$

- Assume  $\text{NH}_3$  and  $\text{H}_2\text{O}$  are the only significant species

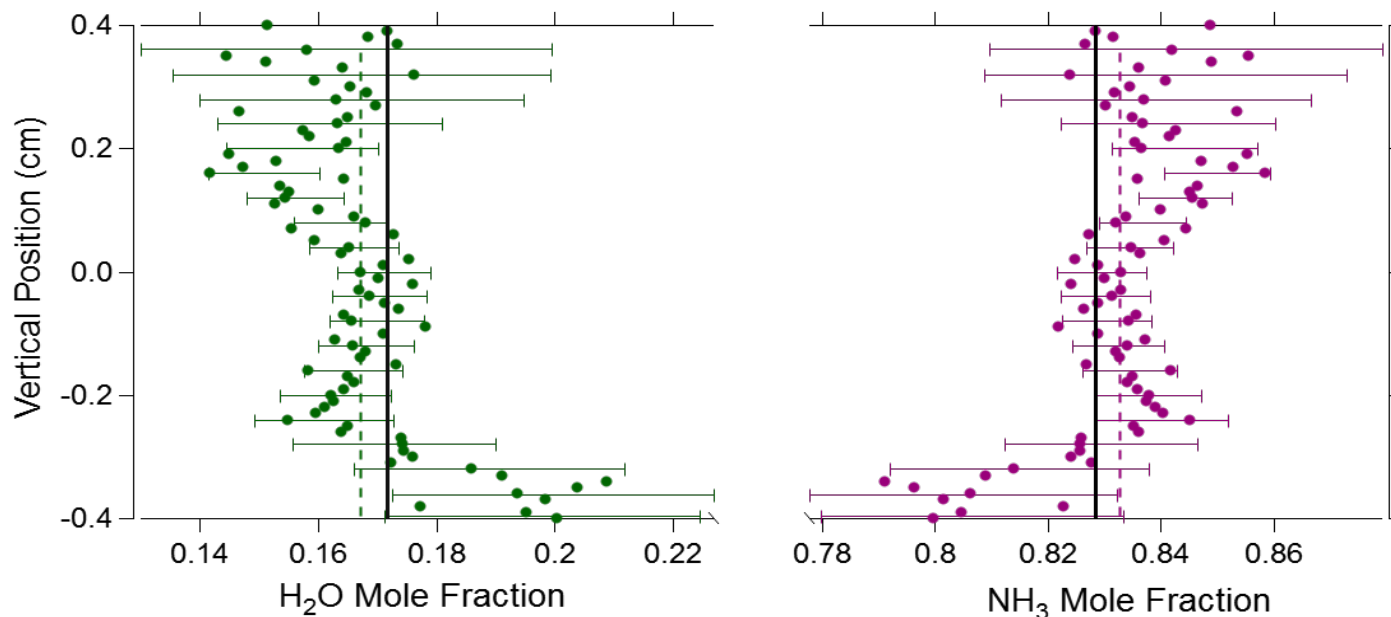
$$\sum_{i=1}^N X_i = 1 \rightarrow X_{\text{NH}_3} + X_{\text{H}_2\text{O}} = 1 \quad (2)$$

- Solve for  $X_{\text{NH}_3}$  by combining Equation 1 and 2

$$X_{\text{NH}_3} = \frac{S_{1388}(T)A_{1527}}{A_{1388}S_{1527}(T) + A_{1527}S_{1388}(T)}$$

# Species Concentration

- **Mean mole fractions from data**
  - $X_{\text{H}_2\text{O}} = 0.167 \pm 0.013$ ,  $X_{\text{NH}_3} = 0.833 \pm 0.013$  (dashed lines)
- **Expected mole fractions based on measured flow rates**
  - $X_{\text{H}_2\text{O}} = 0.172$ ,  $X_{\text{NH}_3} = 0.828$  (solid lines)
- **Uncertainty on edges due to temperature uncertainty in  $S(T)$**





# WMS as a Diagnostic

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## Benefits:

- **Small robust system**
- **Relatively inexpensive**
- **Easy alignment**
- **Calibration free**
- **Measurements of temperature and mole fraction**
- **Low power**
- **Fast acquisition times**
- **Non-intrusive**

## Drawbacks:

- **Path averaged 1D measurements**
- **Velocity measurements are challenging**

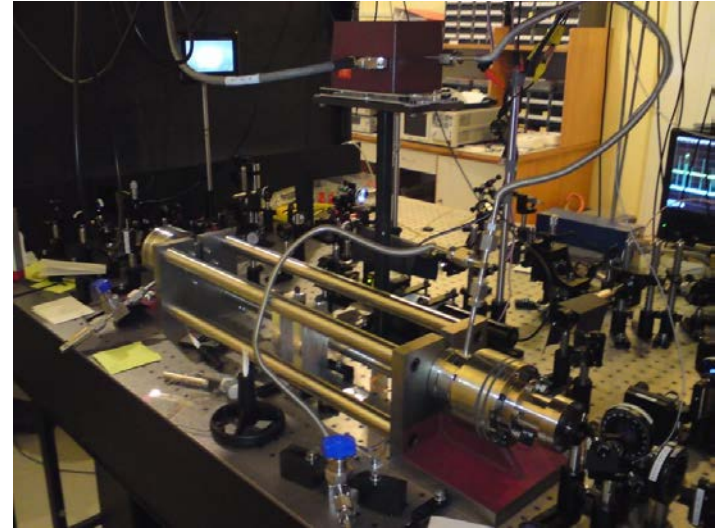
# WMS as a Diagnostic

## Benefits:

- Small robust system
- Relatively inexpensive
- Easy alignment
- Calibration free
- Measurements of temperature and mole fraction
- Low power
- Fast acquisition times
- Non-intrusive

## Drawbacks:

- Path averaged 1D measurements
- Velocity measurements are challenging



**Cavity Ring Down Spectroscopy  
(CRDS)**

<http://ocmq.s.fizyka.umk.pl/?q=node/12>

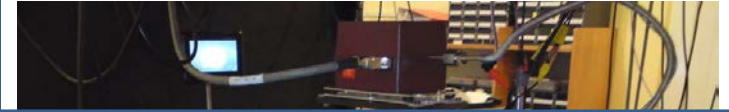
# WMS as a Diagnostic

## Benefits:

- Small robust system
- Relatively inexpensive
- Easy alignment
- Calibration free
- Measurements of temperature and mole fraction
- Low power
- Fast acquisition times
- Non-intrusive

## Drawbacks:

- Path averaged 1D measurements
- Velocity measurements are challenging



**Planar Laser Induced  
Fluorescence (PLIF)**

<http://nccrd.in/facilities/equipment/19>

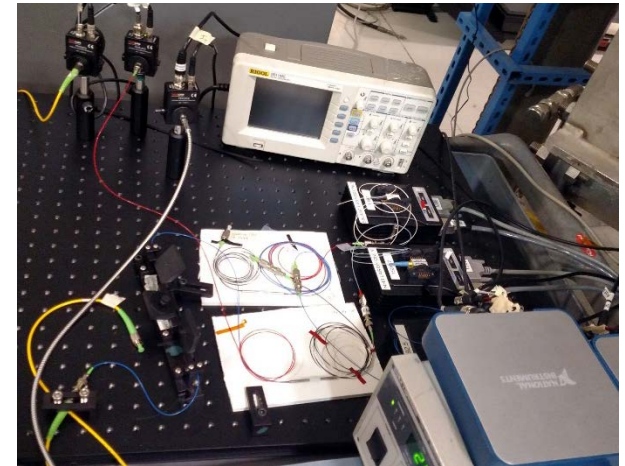
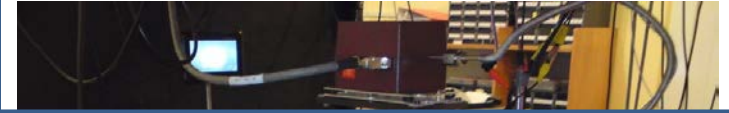
# WMS as a Diagnostic

## Benefits:

- Small robust system
- Relatively inexpensive
- Easy alignment
- Calibration free
- Measurements of temperature and mole fraction
- Low power
- Fast acquisition times
- Non-intrusive

## Drawbacks:

- Path averaged 1D measurements
- Velocity measurements are challenging



**Wavelength Modulation  
Spectroscopy (WMS)**

# Conclusions and Acknowledgements

## Conclusions:

- **Spatial distributions of temperature were calculated within the plume**
- **Radial temperature distributions were calculated using Abel inversions**
  - Temperature distributions consistent with previous experimental results [13,14]
- **New method for calculating species concentration was proposed**
  - Concentration measurements compare well with expected results
- **Results indicate WMS is a promising technique for microthruster diagnostics**

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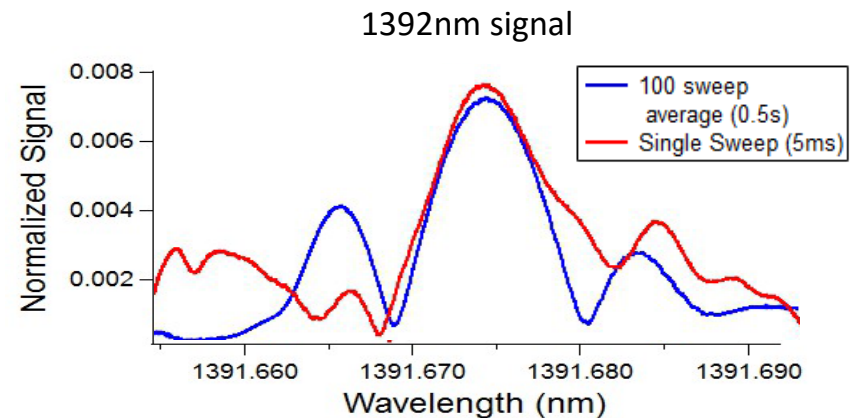
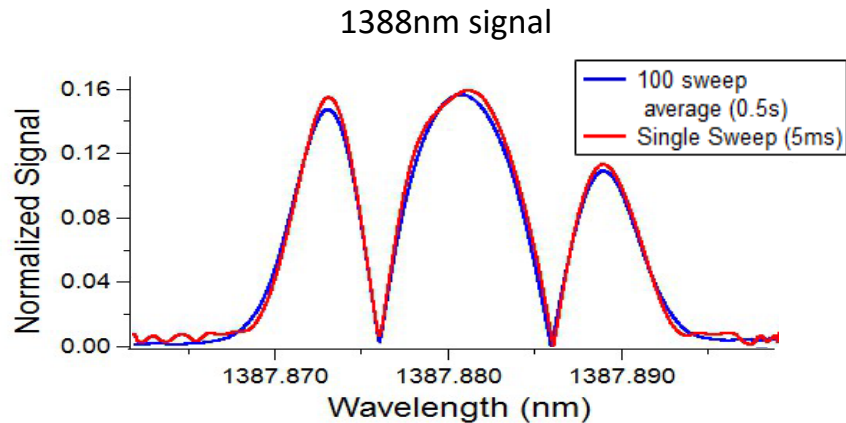
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# Questions?

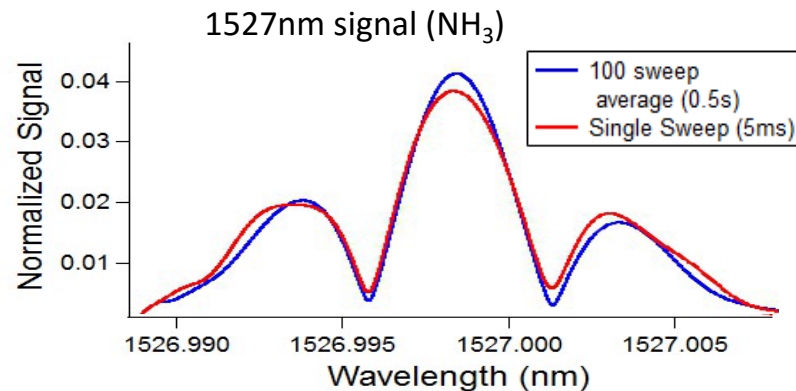


# Data Processing

## Water absorption features



## NH<sub>3</sub> absorption feature

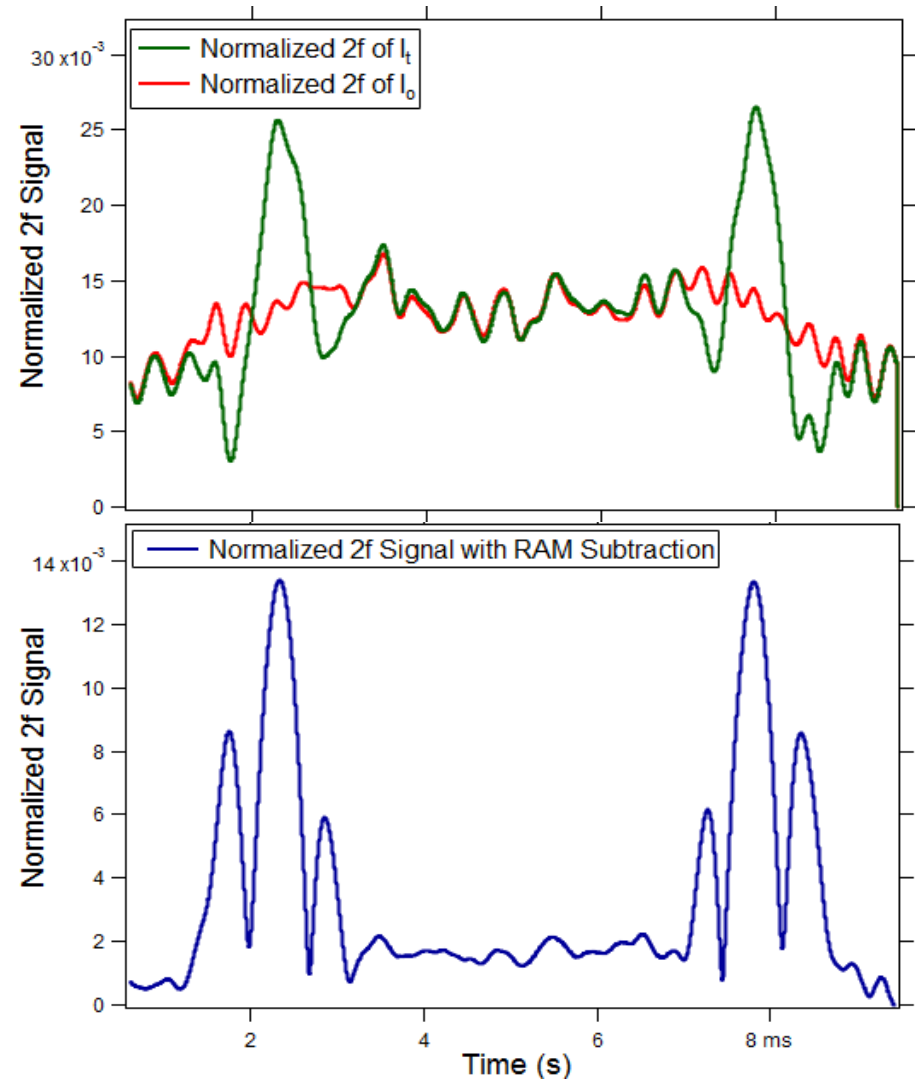


**Data sets were averaged  
for 0.5 seconds to  
increase signal to noise  
ratio (SNR)**



# Residual Amplitude Modulation (RAM)

- Non-linearity in the laser 2 was substantial
- $I_o$  signal was used for background subtraction to account for this

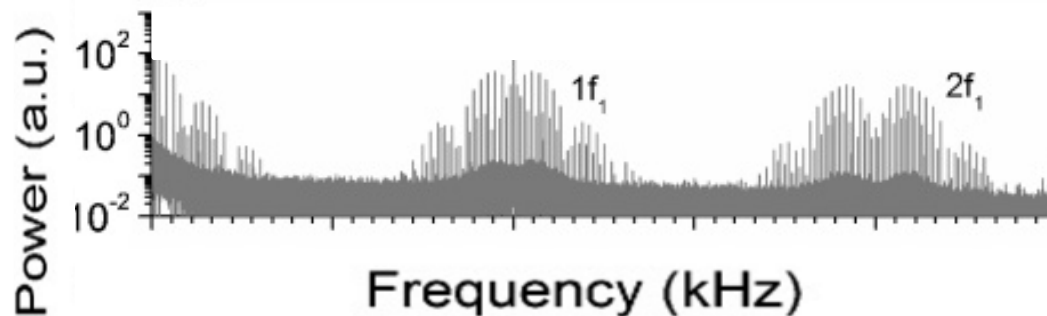


# Wavelength Modulation Spectroscopy

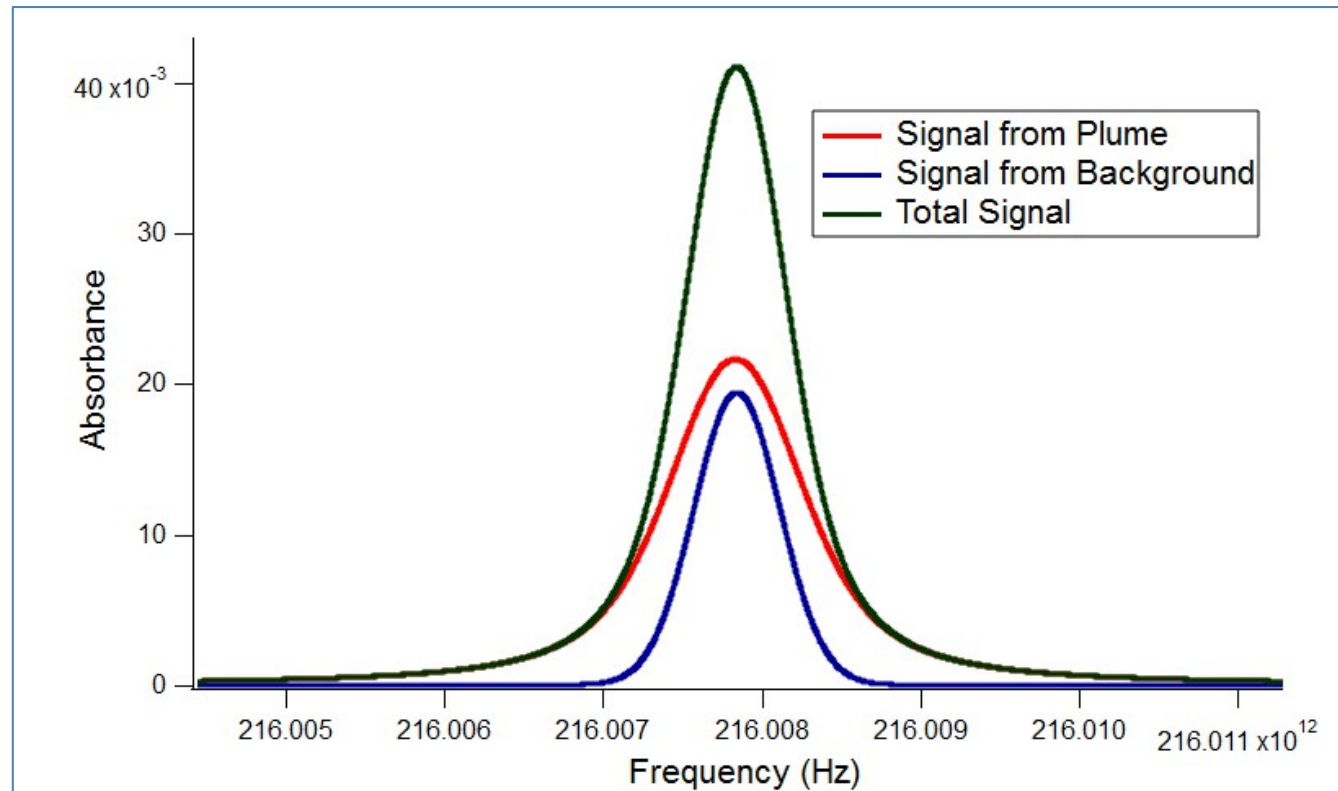
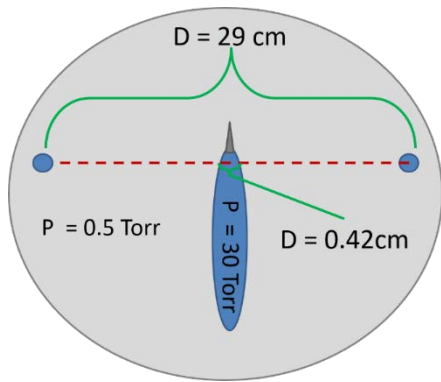
X Component:  $A \sin(fx + \phi) \cos(fx) = \frac{A}{2} (\sin(\phi) + \sin(2fx + b))$

Y Component:  $A \sin(fx + \phi) \sin(fx) = \frac{A}{2} (\cos(\phi) - \cos(2fx + b))$

Magnitude:  $\sqrt{\left(\frac{A}{2} \sin(\phi)\right)^2 + \left(\frac{A}{2} \cos(\phi)\right)^2} = A/2$



# Data Processing



# Inverse Abel Transform

